Attachment 5

Airworthiness Directives and Service Bulletin Status

ASIANA AIRLINES

7. AIRFRAME AD (HL7742 MSN 29171,LN107)

As of 2013.07.23

WCF - Whichever Comes First, WCL - Whichever Comes Later

RE - Repetitve Inspection, N/A - Not Applicable,
SS - Superseded, CC - Cancellation Notice,
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ATA	FAA AD		AD DESCRIPTION	EFFECTIVITY		COMPLIANCE DATE	REALTED DOCUMENT	AD CARD		AAR Action Status
	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			REPETITIVE CARD (Y or N)	INTERVAL	
25	FAA 74-08-09R2	29-Jul-1996	TO PREVENT POSSIBLE FIRES THAT COULD RESULT FROM SMOKING MATERIALS BEING DROPPED INTO LAVATORY OR LINEN WASTE RECEPTACLE	All transport category airplanes, certified in any category, that have one or more lavatories equipped with paper or linen wste receptacles.	B777 ALL	No time limit	FAA 74-08-09 R3			SS See FAA 74-08-09 R3.
55	FAA 96-14-06	24-Jul-1996	Repetitive inspections for broken lockwires on the bearing retainer nut of the pivot fittings of the horizontal stabilizer	Applicable to 777-200 Line # 1, 3, 5, 7, 8, 9, 11, 12, 13, 15, 16, 17, 19, 20, 21, 22, and 23 only.	N/A	-	SB 777-55A0003 R1			N/A
31	FAA 96-20-01	10-Oct-1996	Requires a one-time inspection to determine the serial numbers of various switch modules on the overhead panel and control stand	Model 777-200 series airplanes, line numbers 1 through 85 inclusive	N/A	-	SB 777-31A0019 R4			N/A
27	FAA 97-10-02	21-May-1997	Repetitive corrosion/resistance inspections to measure the resistance of each wire bundle of the flight control system	All Model 777 series airplanes	N/A	-	SB 777-27A0019			N/A
26	FAA 97-10-11	12-May-1997	To prevent damage to the engine fire shutoff switch (EFSS) solenoid and to the override mechanism, and consequent failure of the EFSS	All Model 777 series airplanes	N/A	-	SB 777-26A0012			SS
53	FAA 97-17-02	04-Sep-1997	Repetitive torquing of the bushing retainer nuts of the pivot pins in the horizontal stabilizer hinge assembly to tighten loose nuts to the new torque value	Applicable to 777-200 Line # 3, 5,7-9, 11-13, 15-17 and 19-22 only	N/A	-	SB 777-53-0006			N/A
23	FAA 97-23-16	22-Dec-1997	Replacement of certain overhead electronics units (OEU) of the passenger address and entertainment communication systems with modified OEU's	Applicable to 777-200 Aircraft WA002-WA017, WA076-WA080, WA086-WA089, WA101-WA104, WA116-WA119, WA131-WA136, WA196-WA198, WA171-WA173, and WA206 only.	N/A	-	SB 777-23A0027			N/A
32	FAA 98-02-06 R1	07-Dec-1999	Repetitive visual inspections to determine the presence and condition of the nut and cotter pin of the lock link mechanism on the side struts and drag struts on the main landing gear (MLG)	Applicable to 777-200 Line # 1-40 only	N/A	-	SB 777-32A0015 SB 777-32A0016			N/A
27	FAA 98-13-12 R1	06-Jul-1998	One-time inspection to detect discrepancies of the fasteners that connect the pushrods to the rudder pedal assemblies	Applicable to 737, 767, 747, 757, 777 Line # 2-93, 95-101, 103, 105-113, & 115 only	N/A	-	SB 777-27A0029			N/A
34	FAA 98-14-10	22-Jul-1998	To prevent erroneous localizer deviation provided by faulty ILS receivers	Model 747-400, 757, 767, and 777 series airplanes; equipped with AlliedSignal RIA-35B Instrument Landing System (ILS) receivers, part number (P/N) 066-50006-0101	N/A	-	-			N/A
25	FAA 98-19-25	22-Oct-1998	Modifying the sliding surface of the door 1 left and door 1 right evacuation slide/rafts	Applicable to 777-200 aircraft Line # 2-48, excluding Line # 10, 41, 43, and 47	N/A	-	SB 777-25A0035			N/A
57	FAA 99-04-19	08-Mar-1999	To detect and correct cracking or missing pieces of the cove skin, or undersized seal inserts installed in the spanwise bulb seals, on the outboard leading edge slats on the wings	Model 777 series airplanes, line numbers 1 through 369 inclusive	HL7596/7597	-	SB 777-57A0034			SS
34	FAA 99-10-14	17-Jun-1999	To prevent erroneous localizer deviation provided by faulty ILS receivers	Applicable to aircraft equipped with AlliedSignal RIA-35B (ILS) receivers P/N 066-50006-0101 only.	N/A	-	-			N/A
57	FAA 99-13-05	08-Jul-1999	To detect and correct fatigue cracking of the outboard support of the flaperon	Applicable to 777 Series aircraft Line # 2-64, excluding Line # 41, 43, and 47.	N/A	-	SB 777-57A0008			N/A
55	FAA 99-14-05	02-Aug-1999	To prevent corrosion and possible cracking of the aluminum ribs and brackets of the trailing edges on the empennage	Applicable to 777-200 aircraft Line # 15-17, and 19-33 only.	N/A	-	SB 777-55A0005 R1			N/A
29	FAA 99-15-10	05-Aug-1999	To prevent failure of an engine-driven pump (EDP) supply shutoff valve, which, in the event of an engine fire	Applicable to 777 Series aircraft equipped with EDP shutoff valve P/N S271W741-21, only.	N/A	-	SB 777-29A0022			N/A
53	FAA 99-17-02	17-Sep-1999	To detect and correct wear of the safety spring wear plate doublers on the auxiliary power unit (APU) firewall	Applicable to 777 Series aircraft Line # 1-183 only	N/A	-	SB 777-53A0018 R1			N/A

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	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			REPETITIVE CARD (Y or N)	INTERVAL	
24	FAA 99-25-13 C1	23-Dec-1999	To prohibit dispatch of an airplane with an engine-mounted backup generator having a sheared shaft; and to detect and correct damage to the engine	Model 777-200 and -300 series airplanes equipped with Rolls-Royce Trent 800, General Electric GE90, or Pratt & Whitney PW4000 series turbofan engines	N/A	-	SB 777-SL-24-023-B			N/A
57	FAA 2000-01-14	18-Feb-2000	To prevent corrosion and subsequent fracture of the fuse pins in the main landing gear attachment and support fittings	Applicable to 777 Series aircraft Line # 1-173 only.	N/A	-	SB 777-57A0029 R1			N/A
57	FAA 2000-02-11	02-Mar-2000	To prevent fuel or fuel vapors from entering the passenger and cargo compartments of the airplane in the event of a failure of the primary seal or development of a crack in the wing center section structure	Applicable to 777-200 IGW aircraft Line # 41-91 only.	N/A	-	SB 777-57-0033			N/A
25	FAA 2000-07-08	16-May-2000	To prevent broken tie rods, which could result in the center stowage bins dropping onto the passenger seats below	Applicable to 777 Series aircraft Line # 2-103, 105-119, 121-161, 163-177 and 179-186 only	N/A	-	SB 777-25-0120 R1			N/A
57	FAA 2000-08-15	31-May-2000	To prevent fatigue cracking of the upper wing skin	Applicable to 777 Series aircraft Line # 1-119, excluding Line # 94, 102, 104, and 118	N/A	-	SB 777-57A0022			N/A
53	FAA 2000-11-11	10-Jul-2000	To prevent fatigue cracking of the aft wheel well bulkhead	Applicable to 777-200 aircraft Line # 1-144 only.	N/A	-	SB 777-53A0015 R1			N/A
28	FAA 2000-14-05	28-Jul-2000	To ensure adequate electrical bonding between the wing spar connectors of the fuel quantity indicating system (FQIS)	Applicable to 777 Series aircraft Line # 1-250 only	N/A	-	SB 777-28A0019			N/A
35	FAA 2000-15-16	12-Sep-2000	To prevent failure of the supplemental oxygen system to deliver oxygen to the passengers and flight attendants in the event of decompression	Applicable to 777 Line # 1-083 only	N/A	-	SB 777-35-0005			N/A
31	FAA 2000-16-16	25-Sep-2000	To minimize contamination of the switch contacts and consequent failure of the switches	Applicable to 777-200 Line # 1-80 only.	N/A	-	SB 777-31A0019 R4			N/A
57	FAA 2000-19-08	10-Oct-2000	To detect and correct cracking or missing pieces of the cove skin, or undersized seal inserts installed in the spanwise bulb seals, on the outboard leading edge slats on the wings	Model 777 series airplanes, line numbers 1 through 369 inclusive	HL7596/7597	-	SB 777-57A0034 R2			N/A
32	FAA 2000-23-20	26-Dec-2000	To prevent corrosion of the axle of the main landing gear	Applicable to 777-200 aircraft Line # 7-11, 26, 28 and 33 only	N/A	-	SB 777-32A0024			N/A
26	FAA 2000-23-28	02-Jan-2000	To prevent damage to the engine fire shutoff switch (EFSS) solenoid and to the override mechanism	All Model 777 series airplanes	N/A	-	SB 777-26A0012			N/A
25	FAA 2000-26-04	02-Feb-2001	To prevent potential ignition of the moisture barrier cover of the drip shield	Boeing Model 747,757,767, and 777 series airplanes having the line numbers (L/Ns) listed in Table 1 of this AD; certificated in any categoty. B777 : 2 through 253 inclusive, except 120,219,230,235,242,245,249.	N/A	-	SB 777-25-0164 R0			SS N/A
57	FAA 2001-03-09	23-Mar-2001	Replacement of nuts on the clevis assemblies that support the auxiliary tracks of the inboard leading edge slats	LINE NO. 1~155, B777 AIRCRAFT	N/A	-	SB 777-57-0038 R0			N/A
57	FAA 2001-08-03	08-May-2001	To prevent migration or loss of the wearplate of the upper housing assembly of the forward trunnion of the main landing gear.	Model 777-200 series airplane, as listed in Boeing Alert Service Bulletin 777-57A0011, Revision 1, dated January 25, 2001, certificated in any category.	N/A	-	SB 777-57A0011 R1			N/A
32	FAA 2001-09-02	06-Jun-2001	To prevent stress corrosion cracking and consequent fracture of the aft trunnion of the outer cylinder of the MLG.	Model 777-200 series airplanes; line numbers(L/N) 2 through 29 inclusive.	N/A	-	SB B777-32A0025			N/A
35	FAA 2001-10-14	08-Jun-2001	To find anf fix incorrect installation of the release pin in the generator firing mechanism of the chemical oxygen generator.	Model 737,747,757,767, and 777 series sirplanes equipped with chemical oxygen generators, certificated in any category.	N/A	-	SB 777-35-0008			N/A
25	FAA 2001-11-11	16-Jul-2001	To prevent unrestrained movement of the passenger seats during high forward deceleration of the airplane.	B737/747/777 aircraft Equipped with Passenger seat manufacured by AVIO INTERIORS Co.	N/A	-	SB 777-25-0111 R1			N/A

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	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			REPETITIVE CARD (Y or N)	INTERVAL	
57	FAA 2001-12-10	20-Jul-2001	To prevent fracture of the inboard support structure of the flaperon, which could result in an in-flight loss of the inboard flaperon, structural damage, and consequent reduced controllability of the airplane.	B777-200 AIRCRAFT L/N 1~9	N/A	-	SB 777-57A0036 R0			N/A
53	FAA 2001-18-09	17-Oct-2001	Inspections for cracking of the web of the horizontal and sloping pressure decks of the fuselage and certain stiffener splice angles and stiffener end fittings	777-200 series airplanes, line numbers 001 through 093 inclusive	N/A	-	SB 777-53-0004 R0			N/A
27	FAA 2001-22-13	06-Dec-2001	Replacing the rudder pedal pushrod fasteners for both the captain's and first officer's pedal assemblies with new, improved fasteners	B777: LINE NO.2-103,105-192	N/A	-	SB 777-27A0030 R0			N/A
35	FAA 2002-06-15	17-Apr-2002	To prevent displacement of the passenger/therapeutic oxygen switch	As listed in Boeing Alert Service Bulletin 777-35A0010 (WA001,WA101-104,WB201-203,WB311-327,WB476,WB501-507,WA076-080,WB001-021,WB031-044,WB411-WB413)	N/A	-	SB 777-35A0010 R0			N/A
54	FAA 2002-07-07		To prevent heat damage of the diagonal brace and forward seals of the aft fairing of the strut	Applicable to 777-200 aircraft equipped with GE90 engines, Line # 6-373 only.	N/A	-	SB 777-54A0017			N/A
55	FAA 2002-08-06	03-May-2002	To prevent failure of the pivot fittings of the horizontal stabilizer, which could result in loss of control of the horizontal stabilizer and consequent loss of control of the airplane.	as listed in Boeing Alert Service Bulletin 777-55A0013 LINE NUMBER 1~100	N/A	-	SB 777-55A0013 R0			SS N/A
57	FAA 2002-11-06	10-Jul-2002	To detect and correct cracking or missing pieces of the cove skin, or undersized seal	Model 777 series airplanes, line number 1 through 369 inclusive	HL7596/7597	1.Ini' insp' : Within TTL 8000 FC or 5000 FC from the effective	SB 777-57A0034 R5			N/A
32	FAA 2002-16-11	20-Sep-2002	To prevent breakage of the aft axle pivot pin of the main landing gear(MLG), which could overload the center axle, causing the tires to blow out upon proper extension of the MLG.	LINE NUMBER 1~263 : B777 A/C	N/A	-	SB 777-32-0029 R0			N/A
28	FAA 2002-16-15	26-Sep-2002	To prevent chafing of the fuel quantity indicator system(FQIS) wiring on surrounding structures and systems.	B777 AIRCRAFT (LINE NUMBER 1~266)	N/A	-	SB 777-28-0012 R0 SB 777-28-0016 R0 SB 777-28-0021 R0			N/A
5	FAA 2003-03-03		To prevent ingestion of ice that could cause shutdown of both engines during icing	Applicable to 777-200/-300 Series aircraft equipped with RR Trent-800 engines only	N/A	-	-			N/A
27	FAA 2003-12-01	17-Jul-2003	To prevent uncommanded stabilizer time due to simultaneous failure of two static seals on one stabilizer trim control module(STCM), combined with failure of the automatic shutdown function of the stabilizer trim system.	All model B777 series airplanes Line No.2~266,273	N/A	-	SB 777-27A0047 R0			N/A
57	FAA 2003-19-02	08-Oct-2003	To prevent cracking of the leading edge out board slats, which could result in separation of the cove skin, structural damage or loss of the trailing edge wedge, and consequent reduced controllability of the airplane.	Model 777 series airplanes, line numbers 1 through 412 inclusive	HL7500/7596/7597/7700	1.Ini' insp' : Within 90 days after the effective date of this AD 2.Rep' insp' : Within 100 FC or 400 FH after ini' insp'. 3.Terminating action : Repl' with New Seal Assy'	SB 777-57A0034 R7 FAA AD 2002-11-06			N/A
54	FAA 2003-25-02	24-Jan-2004	To prevent leakage of hydraulic fluid into the strut aft dry bay, where high temperatures associated with the adjacent primary exhaust nozzle may ignite the fluid, resulting in an uncontrolled fire in the strut aft dry bay.	Model 777-200 and 777-300 series airplanes having line numbers 2 through 297 inclusive, 299 and 300	N/A	-	SB 777-54A0016 R0/R1			N/A
53	FAA 2004-03-05	24-Feb-2004	To detect and correct cracks or damage to the web of the aft pressure bulkhead.	B777-200 AIRCRAFT BLOCK NUMBER WA207, WB325	N/A	-	SB 777-53A0039 R0			N/A
22	FAA 2004-04-08	26-Feb-2004	To prevent the possibility of the airplane departing the runway during Category IIIB autoland operations due to autopilot disconnect in low visibility weather condition and to warn the flightcrew of the potential for autopilot disconnect or unscheduled speed brake retraction during any landing.	B777-200 BLOCK NO. WC381-385 WC446 WC447	N/A	-	-			N/A

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25	FAA 2004-05-26	19-Apr-2004	To prevent uncommanded movements of the power drive unit during ground handling of cargo and consequent possible injury to ground personnel.	AS LISTED IN BOEING SB 777-25-0191	HL7596/7597	WITHIN 18 MONTHS AFTER EFFECTIVE DATE OF THIS AD.	SB 777-25-0191 R0			N/A
27	FAA2004-12-15	22-Jul-2004	To prevent damage to the stabilizer cutout circuit wires in the bundles due to contact between the bundles and the adjacent galley water drain tube and hydraulic tubes.	B777-200 airplanes, as listed in Boeing SB 777-27-0057.	N/A	-	SB 777-27-0057 R0			N/A
26	FAA 2004-17-05	07-Oct-2004	To prevnet leakage of fire extinguishing agent through the filter.regulator of the cargo fire extinguishong system.	L/N 002-290 B777-200 airplanes and 777-300	N/A	-	SB 777-26-0028 R0			N/A
26	FAA 2004-18-09	19-Oct-2004	To prevent a possible source of ignition in a flammable leakage zone.	Applicable to 777-200 and 777-300 series airplanes, line numbers 1-400.	N/A	-	SB 777-28-0028 R1			N/A
52	FAA 2005-05-20	22-Apr-2005	SENSITIVE SECURITY AD F/D DOOR - DECOMPRESSION SENSING MODULE CHANGE	SB 747-52-2274: 747-100, 747-100B, 747-100B SUD, 747-200B, 747-200F, 747-300, 747-400, 747-400D, 747SP, 747SR SB 767-52-0087: 767-200, 767-300 SB 777-52-0035: 777-200, 777-300, 777-300ER	HL7500/7596/7597/7700/7732 (Total 5 A/C)	Within 120 Days from Apr 14, 2005.	SB 777-52-0035			N/A
32	FAA 2005-07-02	04-May-2005	To detect and correct cracks or defects that could result in a fracture of the cylinder of the MLG	Specific MLG Outer Cylinder S/N equipped on B777 airplanes (As listed in SB 777-32A0038 R1)	N/A	-	SB 777-32A0038 R1			N/A
25	FAA 2005-07-16	12-May-2005	To prevent a galley, purser work station, or closet from detaching from the tie-down, fitting studs during an emergency landing.	B777-200 and -300 A/C AS LISTED IN SB 777-25-0217.	HL7596	Within 60 Months from effective date of this AD.	SB 777-25-0217 R0			N/A
34	FAA 2005-10-03	15-Jun-2005	To prevent the display of erroneous heading information to the pilot.	B777-200 & -300 SERIES A/C BELOW LN 409 as listed in Boeing Service Bulletin 777-34A0082, Revision 1.	HL7596/7597/7500/7700	WITHIN 6 MONTHS AFTER THE EFFECTIVE DATE OF THIS AD.	SB 777-34A0082 R1 SB 777-34-0094			N/A
57	FAA 2005-10-17	23-Jun-2005	To prevent fatigue cracks in the lower t-chord at the bolt holes commom to the paddle fittings that could result in fractures of one or more of the t-chord segments.	GROUP 1 AIRPLANE : B777-200 L/N 1-240 GROUP 2 AIRPLANE : B777-200 IGW L/N 41-261 GROUP 3 AIRPLANE : B777-300 L/N 94-262	N/A	-	SB 777-57A0040 R2			N/A
28	FAA 2005-10-20	23-Jun-2005	PREVENT FUEL PUMP IN CENTER FUEL TANK FROM RUNNING DRY AND BECOMING A POTENTIAL IGNITION SOURCE.	Boeing Model 77-200 series airplanes, certified in any category; as identified in Boeing Special Attention Service Bulletin 777-28-0036, Block Number : WC111~WC115	N/A	-	SB 777-28-0036 R0			N/A
25	FAA 2005-12-10	20-Jul-2005	PREVENT OVHT OF FREQUENCY CONVERTER'S OUTPUT WIRING	As listed in SB 777-25-0210 R0.	N/A	-	SB 777-25-0210 R0			N/A
28	FAA 2005-13-20	29-Jul-2005	To prevent a latent open circuit that could leave the fuel spar shoutoff valve in a partially open position when the engine fire switch is	MODEL B777-200 and -300 series airplanes, : L/N 1-360	HL7596/7597	Within 29 Jul 2010.	SB 777-28-0025			N/A
25	FAA 2005-13-28	01-Aug-2005	To prevent failure of the vertical tie rods supporting certain electrical racks and the center stowage bins.	B777-200 and -300 AIRCRAFT : L/N 002 - 283	N/A	-	SB 777-25-0144			N/A
27	FAA 2005-13-29	01-Aug-2005	INST' OF THE TIE PLATE FOR WIRE BUNDLES ROUTED to prevent wire chafing.	B777-200 and -300 series airplanes, certificated in any category; as identified in Boeing Alert Service Bulletin 777-27A0060. : L/N 1-419	HL7500/7596/7597/7700	WITHIN 18 MONTHS AFTER EFFECTIVE DATE OF THIS AD.	SB 777-27A0060 R0			N/A
33	FAA 2005-13-34	02-Aug-2005	To prevent overheating of ceiling and sidewall light connectors.	B777-200 and -300 Aircraft L/N 001-264	N/A	-	SB 777-33-0019 R1			N/A
33	FAA 2005-14-04	15-Aug-2005	REPL' ALL HALOGEN LAMPS IN CGO COMP' CEILING L'T to prevent a fire in the cargo compartment.	B777-200 & -300 A/C as identified in Boeing SB 777-33-0025.	HL7596/7597/7500/7700	WITHIN 18 MONTHS AFTER EFFECTIVE DATE OF THIS AD.	SB 777-33-0025 R0			N/A
57	FAA 2005-14-05	10-Aug-2005	To prevent uncontrollable fire in the leading edge of the wing.	B777-200 & -300 A/C as identified in Boeing SB 777-57-0046.	HL7500/7596/7700	WITHIN 18 MONTHS AFTER EFFECTIVE DATE OF THIS AD.	SB 777-57-0046 R0			N/A
28	FAA 2005-17-02	26-Sep-2005	INSP' OF THE V/V CONTROL & IND' WIRE BUNDLES OF FUEL SYSTEM TO PREVENT CHAFING.	B777-200 & -300 A/C as identified in Boeing SB 777-28-0033.	HL7500/7596/7597/7700	WITHIN 18 MONTHS AFTER EFFECTIVE DATE OF THIS AD.	SB 777-28-0033 R0			N/A

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25	FAA 2005-17-09	27-Sep-2005	To prevent potential ignition of the moisture barrier cover of the drip shield	Boeing Model 747,757,767, and 777 series airplanes having the line numbers (L/Ns) listed in Table 1 of this AD; certificated in any categoty. B777 : 2 through 253 inclusive, except 120,219,230,235,242,245,249.	N/A	Within 6 years from effective date of FAA AD 2000-26-04 (02Feb,2000).	SB 777-25-0164 R0			N/A
53	FAA 2005-18-10	19-Oct-2005	To prevent loss of the capability of the cabin floor and seat track structure to supprt the airplane interior inertia loads under emergency landing conditions.	B777-200/300 SERIES AIRPLANES AS IDENTIFIED IN Being Special Attention SB 777-53-0042.	HL7596/7597/7500/7700	Within 19 Oct 2010.	SB 777-53-0042			N/A
34	FAA 2005-18-51	29-Aug-2005	Modification of the OPS of ADIRU from S/W Ver P/N 3470-HNC-100-03 to S/W Ver P/N 3475-HNC-100-06 or 3474-HNC-100-07. to prevent the operational program software(OPS) from using data from faulted sensors.	ALL B777 AIRPLANES	HL7596/7597/7500/7700/7732/7739	WITHIN SEP 03, 2005	SB 777-34A0137 R0			N/A
27	FAA 2005-25-24	20-Jan-2006	To prevent damage and eventual fracture of the yoke assembly, pin assembly, and attachment bolts that connect the inboard and outboard PCUs to a flaperon	Applies to Boeing Model 777-200 and -300 series airplanes, certificated in any category, as identified in Boeing Service Bulletin 777-27A0056, Revision 1, dated July 8, 2004.	N/A	-	SB 777-27A0056 R1			N/A
28	FAA 2006-05-08	13-Apr-2006	To prevent energy from a lightning strike on the bushing for the sump drain valve from arcing to the inside of the center fuel tank wall.	B777-200 series airplanes, certificated in any category; as identified in SB 777-28-0045	N/A	-	SB 777-28-0045			N/A
52	FAA 2006-11-13	30-Jun-2006	To prevent failure of the EPAS, which could result in the inability to open the exit door during an emergency evacuation.	B777-200 and -300 series airplanes, certificated in any category; as identified in SB 777-52-0033, Rev1,	HL7596/7597/7500/7700	Within 24 months after the effective date of this AD.	SB 777-52-0033 R1 SB 285W0955-24-01 (EP 5211-01041)			N/A
25	FAA 2006-12-06	12-Jul-2006	To prevent fire propagation or smoke in the cabin area due to electrical arcing or sparking and ignition of the spiral wire wrapping.	Boeing transport category airplanes equipped with certain Driessen Aircraft Interior Systems or Showa Aircraft Industries galleys, certificated in any category; as identified in paragraphs (c)(1) through (c)(5) inclusive of this AD. (1) Model 737-300, -400, -500, -700, and -800 series airplanes; (2) Model 747-400 and 747-400F series airplanes; (3) Model 757-200 series airplanes; (4) Model 767-300 series airplanes; and (5) Model 777-300 series airplanes.	N/A	-	SB 777-25-0180			N/A
28	FAA 2006-12-26	21-Jul-2006	To prevent arcing or sparking during a lightning strike at the interface between the bulkhead fittings of the engine fuel feed tube and the front spar inside the fuel tank.	B777-200, -300, and -300ER series airplanes, certificated in any category; as identified in SB 777-28-0044, Rev1,	HL7596/7597/7500/7700/7732	Within 60 Months from the Effective Date of this AD. (2006.07.21)	SB 777-28-0044 R1			N/A
26	FAA 2006-17-11	27-Sep-2006	to prevent failure of the safety fittings for the cargo fire extinguishing bottles due to corrosion.	airplanes listed in Table 1 of this AD, certificated in any category; B777-200 and -300 series airplanes;as Identified in SB 777-26-0034, Rev1	HL7596/7597/7500/7700	<Initial Insp' & Rep' Insp'> Within 6,000 FH or 18 Months from the AD effective date,WCF. <Terminating Action> Within 2011.09.27	SB 777-26-0033			N/A
34	FAA 2006-16-18	25-Sep-2006	To prevent a bearing error, which could lead to an airplane departing from its scheduled flight path	Sandel Avionics Incorporated (Sandel) Model ST3400 terrain awareness warning system/radio magnetic indicator (TAWS/RMI) units approved under Technical Standard Order(s) C113, C151a, or C151b; as identified in Sandel ST3400 Service Bulletin SB3400-01, RevB, as installed on various small and transport category airplanes, certificated in any category	N/A	-	SB 3400-01R1			N/A

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ATA	FAA AD		AD DESCRIPTION	EFFECTIVITY		COMPLIANCE DATE	REALTED DOCUMENT	AD CARD		AAR Action Status
	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			REPETITIVE CARD (Y or N)	INTERVAL	
34	FAA 2006-16-18R1	18-Jul-2008	To prevent a bearing error, which could lead to an airplane departing from its scheduled flight path	Sandel Avionics Incorporated (Sandel) Model ST3400 terrain awareness warning system/radio magnetic indicator (TAWS/RMI) units approved under Technical Standard Order(s) C113, C151a, or C151b; as identified in Sandel ST3400 Service Bulletin SB3400-01, RevB, as installed on various small and transport category airplanes, certificated in any category	N/A	-	SB 3400-01, Revision B.			N/A
54	FAA 2006-19-12	30-Oct-2006	To prevent cracking of lower webs of the aft fairings.	B777-200 and -300 series airplanes, certificated in any category; as identified in SB 777-54-0021, Rev 1,	HL7500/7596/7597/7700/7732	Within 12 months after the effective date of this AD.	SB 777-54-0021R1			N/A
21	FAA 2006-21-05	22-Nov-2006	To prevent an increased pressure drop across the humidifier and consequent reduced airflow to the flight deck.	B747-400, 777-200, and 777-300 series airplanes, certificated in any category; as identified in SB 747-21A2414, Rev 3, ; and SB 777-21A0048, Rev3,	N/A	-	SB 747-21A2414 SB 777-21A0048			N/A
53	FAA 2006-25-05	11-Jan-2007	To detect and correct corrosion, and prevent subsequent fatigue cracks, on the fuselage skin under the forward and aft wing-to-body fairings, which could result in rapid decompression of the airplane.	B777-200, -300, and -300ER series airplanes; certificated in any category; as identified in SB 777-53A0044, Rev 1,	HL7596/7597/7500/7700/7732	A. Within 1,500 days after A/C Certification B. Within 1,500 days after the latest Zonal or Surveillance Inspection after the effective date of this AD. C. Within 750 days after the effective date of this AD (Within 2009.01.30) ,WCL.	SB 777-53A0044			N/A
27	FAA 2007-02-23	02-Mar-2007	To prevent disconnection of the drive arm from its drive gimbal.	B777-200, -300, and -300ER series airplanes, certificated in any category; as identified in Alert SB 777-27A0073.	HL7500/7596/7597/7700/7732/7739	Within 24 months after the effective date of this AD.	SB 777-27A0073			N/A
36	FAA 2007-07-05	18-Apr-2007	To prevent an ASCPC failure that could stop airflow into the airplane, inhibit the cabin altitude warning message, and cause an incorrect display of cabin altitude.	B777-200, -200LR, -300, and -300ER series airplanes, certificated in any category	HL7596/7597/7500/7700/ 7732/7739/7742	Within 90 days after the effective date(By 2007.07.17)	SB 777-36A0026			Completed MEO SB 777-36A0026.000.01 2007.03.16 FH/FC : 5,139.75/918 Completed MEO SB 777-36A0026.001.01 2007.05.17 FH/FC : 6,049.73/1,074
36	FAA 2007-07-05 R1	18-Apr-2007	To prevent an ASCPC failure that could stop airflow into the airplane, inhibit the cabin altitude warning message, and cause an incorrect display of cabin altitude.	B777-200, -200LR, -300, and -300ER series airplanes, certificated in any category	HL7596/7597/7500/7700/ 7732/7739/7742	Within 90 days after the effective date(By 2007.07.17)	SB 777-36A0026			Completed MEO SB 777-36A0026.000.01 2007.03.16 FH/FC : 5,139.75/918 Completed MEO SB 777-36A0026.001.01 2007.05.17 FH/FC : 6,049.73/1,074

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ATA	FAA AD		AD DESCRIPTION	EFFECTIVITY		COMPLIANCE DATE	REALTED DOCUMENT	AD CARD		AAR Action Status
	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			REPETITIVE CARD (Y or N)	INTERVAL	
57	FAA 2007-09-04	04-Jun-2007	To detect and correct corrosion or cracking of the torque tube and closeout rib fittings that support the inboard end of the inboard trailing edge flap.	B777-200, -300, and -300ER series airplanes, certificated in any category; as identified in SB 777-57-0054	HL7596/7597/7500/7700/7732	A.Within AD Specified Compliance Time after AD effective date. B. <Initial Insp'> Within Compliance listed in table 7 of SB 777-57A0048 except the following A. C. <Terminating Action> Modification - Within 60 months after the effective date of this AD. (within 2012-06-03) Refer to SB 777-57-0054	SB 777-57-0054			N/A
27	FAA 2007-13-05	25-Jul-2007	To prevent flutter, which can cause damage to the control surface structure and consequent loss of control of the airplane.	B777-200, -200LR, -300, and -300ER series airplanes, certificated in any category	ALL B777 A/C	Refer to this AD. Note (b) 참조	SB 777-27-0062	FAA 2007-13-05.000.00 (GTL 37810)	5,000FC or 16Month WCF	<Initial Insp'> Completed MEO SB 777-27-0062.000.00 2007.08.30 FH/FC : 74777/1315
27								FAA 2007-13-05.001.00 (GTL 37811)	12,000 F/H or 36 Month WCF	 Completed MEO SB 777-27-0062.002.00 2008.11.28 FH/FC : 13890/2283
27								FAA 2007-13-05.002.00 (GTL 37812)		 Completed MEO SB 777-27-0062.001.00 MEO SB 777-27-0062.003.00 2008.11.27 FH/FC : 13890/2283
27								FAA 2007-13-05.003.00 (GTL 37813)		 RE
27	FAA 2007-17-12	02-Oct-2007	To prevent an undetected failure of the primary load path for the ballscrew in the drive mechanism of the horizontal stabilizer and subsequent wear and failure of the secondary load path, which could lead to loss of control of the horizontal stabilizer and consequent loss of control of the airplane.	B777 airplanes, certificated in any category.	HL7596/7597/7500/7700/7732/7739/7742/7755/7756	Refer to this AD. Note (a) 참조	SB 777-27A0059			<Initial Insp'> Completed MEO SB 777-27A0059.000.00 2008.11.28 FH/FC : 13,890.92/2,283 SS
27	FAA 2007-23-11	20-Dec-2007	To prevent wear and corrosion at the flap support joints.	B777-200, -200LR, -300, and -300ER series airplanes, certificated in any category; as identified in SB 777-27A0066, Rev 1	HL7500/7596/7597/7700/7732/7739	Refer to this AD about <Initial Insp'>& <Rep' Insp'>	SB 777-27A0066 R1, SB 777-27A0071 R1			N/A
55	FAA 2007-26-05	22-Jan-2008	To detect and correct a cracked actuator fitting.	B777-200, -200LR, -300, and -300ER series airplanes, certificated in any category	HL7596/7597/7500/7700/7732/7739/7742/7755/7756	Refer to this AD about <Initial Insp'>& <Rep' Insp'>	SB 777-55A0015			SS

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	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			REPETITIVE CARD (Y or N)	INTERVAL	
30	FAA 2008-09-11	06-Jun-2008	To prevent a fire near a composite drain mast and possible disruption of the electrical power system.	B777-200, -300, and -300ER series airplanes, certificated in any category, as identified in SB 777-30-0014,	HL7596/97/7500/7700/7732	Within 60 months after the effective date of this AD. (Within 2013. 06. 05)	SB 777-30-0014 SB 777-38-0026			N/A
28	FAA 2008-11-13	03-Jul-2008	To prevent the potential for ignition sources inside fuel tanks caused by latent ailures, alterations, repairs, or maintenance actions	B777-200, -200LR, -300, and -300ER series airplanes; certificated in any category; with an original standard airworthiness certificate or original export certificate of airworthiness issued before December 5, 2007	B777 ALL	Within 2008.12.16	D622W001-9			AAR Maint' Program Revision
35	FAA 2008-12-05	16-Jul-2008	To prevent the in-line flow indicators of the passenger oxygen masks from fracturing and separating,	B777-200, -200LR, -300, and -300ER series airplanes, certificated in any category; as identified in SB 777-35-0019,	HL7500/7596/7597/7700/7732/7739/7742	Within 60 months after the effective date of this AD.	SB 777-35-0019			Completed MEO SB 777-35-0019.000.01 2010.04.12 FH/FC 20872/3199
26	FAA 2008-14-11	23-Jul-2008	To prevent ETOPS operation with insufficient cargo fire suppression capability	B777-200, -200LR, -300, and -300ER series airplanes, certificated in any category; approved for extended-range twin-engine operational performance standards (ETOPS)	HL7596/7597/7500/7700/7732/7739/7742/7755/7756	Within 1 months after the effective date of this AD.	SB 777-26-0045			Op' Spec' Revision
53	FAA 2008-16-12	09-Sep-2008	to detect and correct wrinkles and cracks in certain external skin panels of Section 48, support structure for the ve	777-200 series airplanes, certificated in any category; as identified in Alert SB 777-53A0051	N/A	-	SB 777-53A0051			N/A
31	FAA 2009-02-05	02-Apr-2009	To prevent an unannunciated loss of cabin pressure. If an undetected loss of pressure event were to cause an unsafe pressure in the cabin	B777-200, -200LR, -300, and -300ER series airplanes, certificated in any category; as identified in SB 777-31A0119 and 777-31A0120, both Rev 2	SB 777-31-0098 SB 777-31-0119: HL7500/7596/7597/HL7700 SB 777-31-0097 SB 777-31-0120 : HL7732/7739/7742	Within 15 months after the effective date of this AD.	SB 777-31A0119 R2 SB 777-31A0120 R2 SB 777-31-0097 R3 SB 777-31-0098 R1			Completed MEO SB 777-31A0120.000.02 2008.10.13 FH/FC : 13,255.94/2,201 Completed MEO SB 777-31-0097.000.03 2008.10.16 FH/FC : 13,298.61/2,205
27	FAA 2009-14-06	12-Aug-2009	To prevent an undetected failure of the primary load path for the ballscrew in the drive mechanism of the horizontal stabilizer and subsequent wear and failure of the secondary load path	Boeing Model 777 airplanes, certificated in any category.	HL7596/7597/7500 HL7700/7732/7739/7742/7755/7756/7775 and after in-service airplane.	Refer to this AD.	SB 777-27A0059 R2			<Initial Insp'> Completed MEO SB 777-27A0059.000.00 2008.11.28 FH/FC : 13,890.92/2,283
27								JIC_FAA 2009-14-06.000.00 (GTL 41227)	3,500 FH or 12 months,WCF.	RE
27								JIC_FAA 2009-14-06.001.00 (GTL 41231)	18,000 FH or 60 months.WCF.	RE
27								JIC_ FAA 2009-14-06.002.00 (GTL 41234)	2,000 FH or 12 months,WCF.	RE
53	FAA 2009-24-08	04-Jan-2010	Inspections for scribe lines in the skin along lap joints, butt joints, certain external doublers, and the large cargo door hinges	B777 A/C as identified in SB 777-53A0054	HL7596/7597/7500/7700/7732/7739/7742/7755/7756	Refer to this AD.	SB 777-53A0054			SS
24	FAA 2009-26-03	01-Feb-2010	To prevent a standby static inverter from overheating	B737/B747/B767/B777 A/C listed in each SB	HL7596/7597/7500/7700/7732/7739	Within 60 months after the effective date of this AD	1.SB 777-24-0095 R1 2.SB 1-002-0102-1000-24-28 Revision A and B			N/A by MSN

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	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			REPETITIVE CARD (Y or N)	INTERVAL	
22	FAA 2010-06-09	01-Apr-2010	To prevent inadvertent engagement of the autopilot during takeoff roll, which could result in rejected takeoff at rotation speed, and consequent possible overrun of the runway	777-200, -200LR, -300, -300ER, and 777F series airplanes as identified in Boeing ASB 777-22A0024.	HL7596/ 7597/ 7500/ 7700/ 7732/ 7739/ 7742/ 7755/ 7756/ 7775	Within Jun 29, 2010 (within 90days After the effective date of this AD)	SB 777-22A0024			Completed MEO SB 777-22A0024.000.00 2010.02.16 FH/FC 20111/3113
57	FAA 2010-13-03	28-Jul-2010	To prevent loss of the lower wing skin load path, which could cause catastrophic structural failure of the wing.	777-200LR and -300ER series airplanes, certificated in any category, as identified in Boeing ASB 777-57A0069.	N/A	-	SB 777-57A0069			N/A
52	FAA 2008-01-01	06-Jul-2010	To prevent failure of this feature, which could jeopardize flight safety	B737/747/767/777 listed in Jamco SB 52-2295 R1/52-2303 R1/52-2302 R1/52-2305 R1	N/A	Within 30 days after the effective date of this AD	Jamco SB 52-2295 R1 SB 52-2303 R1 SB 52-2302 R1 SB 52-2305 R1			N/A
57	FAA 2010-14-13	17-Aug-2010	To detect and correct damage to the outboard slat main track slat cans, which can allow fuel leakage into the fixed wing leading edge in excess of the capacity of the draining system	777-200, -200LR, -300, and -300ER as identified in Boeing Service Bulletin 777-57A0064 R1.	HL7596/7597/7500/7700/7732/7739/7742/7755/7756	WITHIN 6 MONTHS AFTER THE EFFECTIVE DATE OF THIS AD. (2011.02.16)	SB 777-57A0064 R1			Completed MEO SB 777-57A0064.000.00 MEO SB 777-57A0064.001.00 2009.06.17 FH/FC : 16665/2665
30	FAA 2010-15-01	17-Aug-2010	To prevent smoke and fire in the cockpit, which could lead to loss of visibility, and injuries to or incapacitation of the flightcrew	B767/B777 airplanes as listed in SB 767-30-0039 & SB 777-30-0012.	HL7596, 7597, 7500, 7700	1.Ini' Insp' : Within 500 FH From effective date. 2.Rep' Insp' : See this AD Summary or AD Card condition 3.Term' action : Replace No.1 Window in type with Pin/Socket	SB 777-30-0012			N/A
57	FAA 2010-23-15	22-Dec-2010	To detect and correct improperly applied sealant, which could result in the disbonding and displacing of sealant, and consequent fuel leaks.	B767/B777 airplanes as listed in SB 777-57-0063 R1	HL7742/7755/7756	Within 36 Months or 6000 FC from AD effective date	SB 777-57-0063			Completed MEO SB 777-57-0063.000.00 2010.04.17 FH/FC : 20872/3199
57	FAA 2010-24-12	20-Jan-2011	To prevent electrical arcing on the fuel tank boundary structure or inside the main and center fuel tanks	777-200, -300, and -300ER airplanes as listed in SB 777-57A0050/57A0051/57A0057/57A0059	SB 777-57A0050 : HL7596/7597/7500/7700/7732 SB 777-57A0051 : HL7596/7597/7500/7700/7732/7739 SB 777-57A0057 : HL7739 SB 777-57A0059 : HL7596/7597/7500/7700/7732/7739/7742/7755/7756	Within Jan 19, 2016	SB 777-57A0050 SB 777-57A0051 SB 777-57A0057 SB 777-57A0059			Completed MEO SB 777-57A0059.000.00 MEO SB 777-57A0059.001.00 2010.04.12 FH/FC 20872/3199 SS
35	FAA 2011-04-09 (Security)	11-Feb-2011	Removing all chemical oxygen generators in the lavatories, or activating all oxygen generators until the generator oxygen supply is expended	Operating under 14 CFR part 121/U.S.-registered and operating under 14 CFR part 129, with a maximum passenger capacity of 20 or greater.	AAR ALL FLEET EXCEPT CARGO	Within Apr 04, 2011	None (PER SSAD)			Completed MEO CB1-25-0007.000.00 2011.02.23 FH/FC 25304/3779
55	FAA 2011-05-12	14-Apr-2011	To detect and correct discrepancies of the HSTA attachment locations	777-200, -200LR, -300, and -300ER as identified in Boeing ASB 777-55A0017	All AAR B777	Within Total 32000 FC or 24 Months from SB issued date, WCL.	SB 777-55A0017	To be issued		Open MEO SB 777-55A0017.000.00

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	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			REPETITIVE CARD (Y or N)	INTERVAL	
28	FAA 2011-09-05	26-May-2011	To prevent potential ignition sources inside fuel tanks	777-200, -300, and -300ER as identified in Boeing Service Bulletin 777-28A0047 R5	HL7732/39/42	Within May 26, 2014	SB 777-28A0040 Rev.1 SB 777-31-0097 Rev.3 GE Aviation SB 5000ELM-28-456 Rev.1 GE Aviation SB 6000ELM-28-457 Rev.1			Completed MEO SB 777-28A0047.001.03 2010.04.19 20872/3199 FH/FC
28										Completed MEO SB 777-28A0040.000.00 2010.04.14 FH/FC 20872/3199
28										Completed MEO SB 777-31-0097.000.03 2008.10.16 FH/FC 13298/2205
28										Completed MEO SB 6000ELM-28-457.000.00 2010.04.14 FH/FC 20872/3199 & MEO SB 6000ELM-28-457.000.01 2010.11.15 FH/FC 23840/3599
28										Completed MEO SB 5000ELM-28-456.000.00 2010.04.12 FH/FC 20872/3199 & MEO SB 5000ELM-28-456.000.01 2010.11.15 FH/FC 23840/3599
25	FAA 2011-07-05	06-May-2001	General visual inspection for cracking of backrest links	Sicma Aero Seats 9140, 9166, 9173, 9174, 9184, 9188, 9196, 91B7, 91B8, 91C0, 91C2, 91C4, 91C5, and 9301 series passenger seat assemblies	N/A	-	SB 90-25-012 ISSUE 5			N/A
54	FAA 2011-09-11	06-Jun-2011	To detect and correct hydraulic fluid contamination, which can cause cracking of titanium parts in the system disconnect assembly	777-200 and -300 series airplanes as identified in SB 777-54A0024, R1	HL7596/7597/7500/7700/7732/7739/7742/7755/7756/7775	1.INI INSP : Within Jun 05 2012 2.REP INSP : Every 6000 FC or 750 Days, WCF 3.TER ACT : Within 6000 FC or 750 Days after detection of Hydraulic Contaimination or Heat Dicoloration, WCF	SB 777-54A0024 R1			Completed Terminating Action MEO SB 777-54A0024.000.00 2010.04.18 FH/FC 20872/3199 & MEO SB 777-54A0024.001.00 2010.04.19 FH/FC 20872/3199
								FAA 2011-09-11.000.00 (GTL 44951)	6000 FC or 750 Day, WCF	N/A

ASIANA AIRLINES

7. AIRFRAME AD (HL7742 MSN 29171,LN107)

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	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			REPETITIVE CARD (Y or N)	INTERVAL	
								FAA 2011-09-11.001.00 (GTL 44952)	6000 FC or 750 Day, WCF	N/A
28	FAA 2011-09-15	06-Jun-2011	To prevent potential ignition sources inside fuel tanks,which, in combination with flammable fuel vapors, could result in a fuel tank explosion and consequent loss of the airplane.	B777-200/-200LR,-300,-300ER as identified in SB 777-28A0038 Rev.1 & SB 777-28A0037 Rev.2	1.SB 777-28A0037 : HL7500, HL7596, HL7597, HL7700 2. SB 777-28A0038 : HL7500, HL7596, HL7597, HL7700, HL7732, HL7739, HL7742, HL7755, HL7756	1.SB 777-28A0037 : Within 2014.06.06 2.SB 777-28A0038 : Within 2016.06.06	SB 777-28A0037 R2 SB 777-28A0038 R0, R1			Completed MEO SB 777-28A0038.000.00 2010.04.19 FH/FC 20872/3199
25	FAA 2011-12-01	01-Aug-2011	To prevent accelerated fires and injuries to the flightcrew and passengers (KOITO SEAT)	Koito Industries, Ltd., seats and seating systems having a model number identified in table 1 of this AD	N/A	-	None			N/A
24	FAA 2011-21-03	16-Nov-2011	To prevent FOD from entering the primary and secondary external power connectors	777-200, -200LR, -300, and -300ER series as identified in SB 777-24-0102 R1 & SB 777-29-0032	SB 777-24-0102: HL7500/7596/7597/7700/7732/7739	Within Nov 15, 2014	SB 777-24-0102 SB 777-29-0032			Completed MEO SB 777-29-0032.000.00 2008.01.15 FH/FC 9398/1637
29					SB 777-29-0032: HL7500/7596/7597/7700/7732/7739/7742					
57	FAA 2011-26-03	03-Jan-2012	To prevent electrical arcing on the fuel tank boundary structure or inside the main and center fuel tanks	777 Airplanes	SB 777-57A0050 : HL7596/7597/7500/7700/7732 SB 777-57A0051 HL7596/7597/7500/7700/7732/7739 SB 777-57A0057 HL7739 SB 777-57A0059 HL7596/7597/7500/7700/7732/7739/7742/7755/7756 FAA 2010-24-12	Within Jan 19, 2016	SB 777-57A0050 SB 777-57A0051 SB 777-57A0057 SB 777-57A0059 FAA 2010-24-12			Completed MEO SB 777-57A0059.000.00 MEO SB 777-57A0059.001.00 2010.04.12 FH/FC 20872/3199
25	FAA 74-08-09 R3	28-Mar-2012	TO PREVENT POSSIBLE FIRES THAT COULD RESULT FROM SMOKING MATERIALS BEING DROPPED INTO LAVATORY OR LINEN WASTE RECEPTACLE	All transport category airplanes, certified in any category, that have one or more lavatories equipped with paper or linen wste receptacles. MODEL : 737,747,767,777, A310,A320,A330 SERIES A/C.	B777 ALL	No time limit	-	FAA 74-08-09 R3.000.00 (GTL 26941)	1000 FH	RE
00	FAA 2012-07-06	15-May-2012	To ensure that fatigue cracking of various principal structural elements (PSEs) is detected and corrected	777-200, -200LR, -300, -300ER, and 777F series airplanes, certificated in any category, with an original airworthiness certificate or original export certificate of airworthiness issued before September 1, 2010	AAR All B777	Within 12 months after the effective date of this AD	MPD Document			Incoroprated into AAR Main't Program
	FAA 2012-08-09	29-May-2012	To detect and correct cracking in the WCS spanwise beams, which could result in reduced structural integrity of the wings	777-200, -200LR, -300, -300ER, and 777F series airplanes as identified in SB 777-57A0087 R1	HL7596/7597/7500/7700/7732/7739/7742/7755/7756/7775/7791	Within 6000 F/C or 1125 Days from AD effective date, WCF or Within total 8000 F/C, WCL	SB 777-57A0087 R1			Open Initial Inspection MEO SB 777-57A0087.000.01

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ATA	FAA AD		AD DESCRIPTION	EFFECTIVITY		COMPLIANCE DATE	REALTED DOCUMENT	AD CARD		AAR Action Status
	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			REPETITIVE CARD (Y or N)	INTERVAL	
57								FAA 2012-08-09.000.00 (GTL 46998)	8000 F/C	RE
55	FAA 2012-08-13	29-May-2012	To prevent failure of the bonding jumper bracket, which could result in loss of lightning protection ground path	777-200 and -300 series airplanes as identified in SB 777-55A0014 R1	N/A	-	SB 777-55A0014 R1			N/A
52	FAA 2012-09-14	21-Jun-2012	To detect and correct fractured and missing latch pin retention bolts in potential separation of the cargo door	777-200, -200LR, -300, -300ER, and 777F as identified in SB 777-52A0038 R1	HL7596/7597/7500/7700/7732/7739/7742/7755/7756/7775	1.Insp' : Within 12 Months from AD effective date. 2.Terminating Action : Before next flight.	SB 777-52A0038 R1			Completed MEO SB 777-52A0038.000.01 2010.10.10 FH/FC 23354/3533
55	FAA 2012-10-10	29-Jun-2012	To prevent a fractured horizontal stabilizer pivot pin, which may cause excessive horizontal stabilizer freeplay and structural damage	777-200, -200LR, -300, -300ER, and 777F series airplanes, as identified in SB 777-55A0018 R1	HL7596/7597/7500/7700 HL7732/7739/7742/7755/7756/7775/7791	1.Pivot Pin Repl' : Within 2018.06.18 2.Rep Insp' : 32000 FC or 6000 Day, WCF from 1.	SB 777-55A0018 R1			Open MEO SB 777-55A0018.000.02
57	FAA 2012-11-03	09-Jul-2012	To prevent an incorrect emergency landing MLG Break-away sequence in puncturing of the wing box and consequent fuel leaks and airplane fire	777-200, -200LR, -300, -300ER, and 777F series airplanes, as identified in SB 777-57A0090	HL7596/7597/7500/7700/7732/7739/7742/7755/7756/7775/7791	Within 2015.08.08	SB 777-57A0090			Open MEO SB 777-57A0090.000.01 MEO SB 777-57A0090.001.01
35	FAA 2012-11-09	10-Aug-2012	To eliminate this hazard and ensure that all lavatories have a supplemental oxygen supply.	(1) Airplanes that are in compliance with the requirements of AD 2011-04-09, Amendment 39-16630 (76 FR 12556, March 8, 2011). (2) Airplanes equipped with any chemical oxygen generator installed in any lavatory and are: (i) Operating under 14 CFR part 121; or (ii) U.S.-registered and operating under 14 CFR part 129, with a maximum passenger capacity of 20 or greater	AAR ALL FLEET EXCEPT B747 Airplane & B767 CARGO	Within 2015.09.09	FAA 2011-04-09			Open
32	FAA 2012-12-08	30-Jul-2012	To prevent structural damage to the side and drag brace lock assemblies, which could result in landing gear collapse during touchdown, rollout, or taxi	777-200 and -S300 series airplanes; certificated in any category; as identified in SB 777-32-0083, Revision 1	N/A	-	SB 777-32-0083 R1			N/A
25	FAA 2012-12-19	30-Jul-2012	To prevent the forward lowered ceiling panels and support structure from becoming dislodged during a 9.0 g forward load and consequent injury to personnel or interference with an emergency evacuation.	777-200, -200LR, and -300ER series as identified in Boeing SB 777-25-0482, Revision 1	N/A	-	SB 777-25-0482 R1			N/A
35	FAA 2012-13-05	16-Aug-2012	To prevent electrical current from passing through the low-pressure oxygen hose internal anti-collapse spring	777-200, -200LR, -300, -300ER, and 777F series airplanes	HL7596/7597/7500/7700/7732/7739/7742/7755/7756/7775/7791	Within 2014.02.15	SB 777-35A0027 R1			Completed MEO SB 777-35A0027.000.01 2012.12.06 FH/FC 34293/4971
32	FAA 2012-12-08C	30-Jul-2012	To prevent structural damage to the side and drag brace lock assemblies, which could result in landing gear collapse during touchdown, rollout, or taxi	777-200 and -S300 series airplanes; certificated in any category; as identified in SB 777-32-0083, Revision 1	N/A	-	SB 777-32-0083 R1			N/A

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ATA	FAA AD		AD DESCRIPTION	EFFECTIVITY		COMPLIANCE DATE	REALTED DOCUMENT	AD CARD		AAR Action Status
	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			REPETITIVE CARD (Y or N)	INTERVAL	
53	FAA 2012-14-03	27-Aug-2012	To detect and correct such fatigue cracking, which could grow large and cause sudden decompression and the inability to sustain limit flight and pressure loads.	777-200 and -300 series airplanes as identified in SB 777-53A0043, L/N 1 - 122	N/A	-	SB 777-53A0043			N/A
32	FAA 2012-19-10	07-Nov-2012	To detect and correct cracking in the MLG center axle and shock strut inner cylinder lugs (pivot joint), which could result in fracture of the MLG pivot joint components and consequent collapse of the MLG.	777-200, -200LR, -300, -300ER, and 777F series airplanes, certificated in any category, as identified in Boeing Alert Service Bulletin 777-32A0082	G1 : HL7597/7500/7700/7732/7739/7742/7755/7756/7775 G2 : HL7596/7791	Refer to AD Summary.	SB 777-32A0082 R0			Completed Initial Inspection MEO SB 777-32A0082.000.00 MEO SB 777-32A0082.001.00 2011.01.24 FH/FC 24866/3729
										Completed Initial Inspection MEO SB 777-32A0082.002.00 MEO SB 777-32A0082.003.00 2012.12.05 FH/FC 34293/4971
								FAA 2012-19-10.000.00 (GTL 48053)	25 Days or 50 FC, WCF	RE
								FAA 2012-19-10.001.00 (GTL 48054)	25 Days or 50 FC, WCF	RE
27	FAA 2012-21-10	14-Nov-2012	To prevent excessive freeplay in the rudder control surface, which could cause rudder vibration, and result in structural damage severe enough to prevent continued safe flight and landing.	777-200LR and -300ER series airplanes as listed in SB 777-27A0109	N/A	-	SB 777-27A0109			N/A
24	FAA 2012-23-06	31-Dec-2012	To prevent failure of the clamp, which could result in wire chafing and potential arcing and consequent fire in section 48 (a flammable fluid leakage zone) or heat damage to APU power feeder cable, insulation blankets, or pressure bulkhead.	777-200, -200LR, -300, -300ER, and 777F series airplanes, certificated in any category, as identified in SB 777-24A0119	HL7596 HL7597 HL7500 HL7700 HL7732 HL7739 HL7742 HL7755 HL7756 HL7775 HL7791	Within 2016.12.31	SB 777-24A0119			Completed MEO SB 777-24A0119.000.00 2012.12.05 FH/FC 34293/4971
28	FAA 2013-05-03	25-Apr-2013	To prevent electrical current from flowing through a motor-operated valve (MOV) actuator into a fuel tank	777-200, -200LR, -300, and -300ER series as identified in Boeing Service Bulletin 777-28A0034 R2.	HL7596 HL7597 HL7500 HL7700 HL7732 HL7739	Within 2018.04.24	SB 777-28A0034 R2			N/A by MSN
24	FAA 2013-05-05	25-Apr-2013	To prevent contactor failures, which could result in uncontained hot debris flow due to ELMS contactor breakdown, consequent smoke and heat damage	777-200, -200LR, -300, and -300ER series as identified in Boeing Special Attention SB 777-24-0106 & 24-0112 R2.	HL7596 HL7597 HL7500 HL7700 HL7732 HL7739 HL7742 HL7755 HL7756 HL7775	SB 24-0106 : Within 2016.04.25 SB 24-0112 : Within 2018.04.25	SB 777-24-0106 SB 777-24-0112 R2			Completed MEO SB 777-24-0106.000.00 2010.04.14 FH/FC 20872/3199
										Open MEO SB 777-24-0112.000.02

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ATA	FAA AD		AD DESCRIPTION	EFFECTIVITY		COMPLIANCE DATE	REALTED DOCUMENT	AD CARD		AAR Action Status
	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			REPETITIVE CARD (Y or N)	INTERVAL	
25	FAA 2013-05-10	03-May-2013	To detect and correct corrosion in the packboard release mechanisms which could interfere with escape slide/raft deployment.	777-200, -200LR, -300, -300ER, and 777F as identified in SB 777-25-0507	B777 Escape Slide	Within 2016.11.03	SB 777-25-0507 VSB 107-25-30			Partially Completed MEO 107-25-30 L1 & R4 : Open L2 : 2012.03.22 L3 : 2011.05.20 L4 : 2011.07.29 R1 : 2011.06.17 R2 : 2012.02.24 R3 : 2011.06.23
53	FAA 2013-07-11	20-May-2013	To detect and correct scribe lines which can develop into fatigue cracks in the skin	B777 A/C as identified in SB 777-53A0054 R1	HL7596/7597/7500/7700/7732/7739/7742/7755/7756	Refer to AD Summary.	SB 777-53A0054 R1			Open MEO SB 777-53A0054.000.01 MEO SB 777-53A0054.001.01 MEO SB 777-53A0054.002.01 MEO SB 777-53A0054.003.01
55	FAA 2013-08-02	24-May-2013	To detect and correct a cracked actuator fitting or incorrectly installed bolts to the actuator fitting	777-200, -200LR, -300, and -300ER series as identified in SB 777-55A0016 R1	HL7596/7597/7500/7700/7732/7739/7742/7755/7756	Refer to AD Summary.	SB 777-55A0015 R2 SB 777-55A0016 R1			<Initial Insp'> Open MEO SB 777-55A0015.000.02 MEO SB 777-55A0015.001.02 <Terminating> Open MEO SB 777-55A0016.000.01 MEO SB 777-55A0016.001.01
35	FAA 2013-08-09	23-May-2013	To prevent the oxygen system flex line from separating from the hard line	777-200, -200LR, -300, -300ER, and 777F as identified in SB 777-35-0024	N/A	-	SB 777-35-0024			N/A per AAR 777 A/C Configuration
25	FAA 2013-11-04	09-Jul-2013	To prevent the failure of urethane seals to maintain sufficient Halon concentrations in the cargo compartments to extinguish or contain fire or smoke.	777-200, -200LR, -300, and -300ER series airplanes, as identified in Boeing Special Attention Service Bulletin 777-25-0362	HL7596/7597/7500/7700/7732	Within 2019.07.08	SB 777-25-0362			N/A by this MSN
54	FAA 2013-11-14	19-Jul-2013	To detect and correct hydraulic fluid contamination of the strut forward dry bay	777-200 and -300 series airplanes equipped with PW4000 series engines as identified in SB 777-54-0028, dated May 25, 2012.	HL7596/7597/7500/7700/7732/7739/7742/7755/7756/7775/7791/8254	1.Initial : Within 600 FC or 12 Months from AD effective date, WCF 2.Repeat : 1200 FC	SB 777-54-0028			Completed Initial Inspection MEO SB 777-54-0028.000.00 MEO SB 777-54-0028.001.00 2013.03.13 FH/FC : 35572/5159
54								FAA 2013-11-14.000.00 - LH (GTL 49686)	1200 FC	RE

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ATA	FAA AD		AD DESCRIPTION	EFFECTIVITY		COMPLIANCE DATE	REALTED DOCUMENT	AD CARD		AAR Action Status
	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			REPETITIVE CARD (Y or N)	INTERVAL	
54								FAA 2013-11-14.001.00 - RH (GTL 49688)	1200 FC	RE

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7. HL7742 Appliance AD

As of 23.Jul.2013

ATA	FAA AD		AD DESCRIPTION	EFFECTIVITY		COMPLIANCE	RELATED DOCUMENT	HL7742 Appliance
	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			
34	FAA 88-25-01	20-Dec-1988	Modify the KDM 7000B DME.	MODEL 737,747 SERIES A/C,EQUIPPED WITH MODEL KDM 7000B DISTANCE MEASURING EQUIPMENT(DME),P/N 066-1019-21/-23/-26/-33/-43/-57,(S/N 21855 - 22526)	N/A	-	KDM 7000B-34-14	Not applicable to P/N installed.
34	FAA 90-05-09	04-Apr-1990	To prevent navigation errors when using the Great Britain (GBR) radio station	TRACOR AEROSPACE, INCORPORATED TA-7800 Omega/VLF Navigation Systems, with Receiver Processor Unit (RPU)	N/A	-	-	Not applicable to P/N installed.
25	FAA 90-09-08	28-May-1990	To prevent failure of the seat back structure	IPECO TSO-C39a CREW SEATS	N/A	-	SB A001-25-29	Not applicable to P/N installed.
34	FAA 92-11-09	28-May-1992	Modify the transponder software & hardware.	1.TPR-720 Air Traffic Control Radar Beacon System/Mode S Transponders, part numbers (P/N) 622-7878-020, and P/N 622-7878-120 2.AAR B747 : N/A	N/A	-	Collins Service Bulletin (SB) TPR-720-34-08 TPR-720-34-C	Not applicable to P/N installed.
35	FAA 92-16-15	01-Sep-1992	Inspection to detect the presence of an over-sized diameter of the probe portion on the oxygen mask plug-in connector.	Scott Aviation Oxygen Mask Plug-In Connectors, Part Number 289-57, as installed B737,747,767 SERIES A/C.	N/A	-	Scott Aviation Service Bulletin 289-35-15	Not applicable to P/N installed.
34	FAA 93-04-02	26-Mar-1993	To prevent a mid-air collision or a near-miss situation caused by failure of these Mode S transponders	ROCKWELL INTERNATIONAL/COLLINS AVIATION DIVISION TDR-94D Mode S Transponders, P/N CPN 622-9210-002	N/A	-	TDR-94/94D-34-6 R2	Not applicable to P/N installed.
26	FAA 93-09-07	11-May-1993	To prevent failure of a fire extinguisher to discharge extinguishant in the event of a fire	PACIFIC SCIENTIFIC COMPANY, HTL/KIN-TECH DIVISION Fire Extinguishers, as listed in Pacific Scientific Alert Service Bulletin 26A1106, dated March 10, 1993	N/A	-	26A1106	Not applicable to P/N installed.
24	FAA 93-12-04	26-Jul-1993	To prevent the presence of smoke in the cockpit, which could prompt the pilot to initiate an emergency landing	Precise Flight, Inc., pulselite units, Model 1210-2405-2; serial numbers X00150 through X01371, inclusive; as installed in various small airplanes in accordance with Supplemental Type Certificate (STC)	N/A	-	PL9303001	Not applicable to P/N installed.
34	FAA 93-23-01	07-Dec-1993	To prevent explosions of galley water heaters and coffee makers and subsequent injuries to passengers or cabin crew members	1.Nordskog water heaters and coffee makers, as listed in Nordskog Industries, Inc., Service Bulletin SB-93-34, dated October 21, 1993; as installed in, but not limited to Boeing Model 727, 737, 747, 757, and 767 series airplanes, Airbus Industrie Model A3	N/A	-	SB 93-34	Not applicable to P/N installed.
35	FAA 93-24-16	04-Feb-1993	To prevent failure of a PBE unit because of a deteriorated neck seal	1.CREWMEMBER PROTECTIVE BREATHING EQUIPMENT (PBE),119003 AND 119003-01 UNITS. 2.CREW PROTECTIVE BREATHING EQUIPMENT MANUFACTURED BY PURITAN BENNETT AERO SYSTEMS Co.	N/A	-	119003-35-1	Not applicable to P/N installed.
34	FAA 94-01-04	04-Feb-1994	To prevent collisions or near misses caused by incompatibility between the TCAS II processors and the current air traffic control system	1.TRAFFIC ALERT AND COLLISION AVOIDANCE SYSTEM (TCAS) II COMPUTER UNITS;P/N 4066010-901,-902,-903 2.HONEYWELL TCAS II COMPUTER	N/A	-	4066010-34-SW16	Not applicable to P/N installed.
34	FAA 94-01-05	04-Feb-1994	To prevent collisions or near misses caused by incompatibility between the traffic alert and collision avoidance system (TCAS) II processors and the current air traffic control system	1.TRAFFIC ALERT AND COLLISION AVOIDANCE SYSTEM II PROCESSORS. 2.EQUIPPED WITH ALLIEDSIGNAL AERODPACE TCAS II PROCESSOR	N/A		-	Not applicable to P/N installed.

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7. HL7742 Appliance AD

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ATA	FAA AD		AD DESCRIPTION	EFFECTIVITY		COMPLIANCE	RELATED DOCUMENT	HL7742 Appliance
	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			
34	FAA 94-02-02	21-Feb-1994	To prevent improper operation of these DME units, which could result in navigational errors	1.DME-700 DISTANCE MEASURING EQUIPMENT(ALL SERIAL NUMBER)(P/N 622-4540-020,622-4540-021,622-4540-022,622-4540-120,622-4540-121) THAT IS INSTALLED ON A/C. 2.EQUIPPED WITH COLLINS DME-700(P/N 622-4540-020/021/022/120/121 DME)	N/A	-	DME-700-34-20 R1	Not applicable to P/N installed.
35	FAA 94-06-04	23-May-1994	To prevent failure of the pin securing the mask shell to the face piece, which could result in a mask leaking oxygen and the crew losing consciousness	1.EROS series MF10-[ ]-[ ] full face quick donning mask regulators installed on but not limited to Airbus A320, Boeing 747-400 2.AAR:HL7413/14/15/16/17	N/A	-	MF10-35-46 R1	Not applicable to P/N installed.
25	FAA 94-14-03	01-Aug-1994	To prevent explosions of galley water heaters and coffee makers, and subsequent injuries to passengers or cabin crew members	Nordskog water heaters and coffee makers, as listed in Nordskog Industries, Inc., Service Bulletin SB-93-35, dated October 21, 1993; as installed in, but not limited to, Boeing Model 727, 737, 747, 757, and 767 series airplanes & Airbus Industrie Model A3	N/A	-	SB-93-35	Not applicable to P/N installed.
25	FAA 94-21-06	25-Nov-1994	To prevent the inability of passengers or crew to egress from their seats during an emergency situation, due to problems associated with the lap belt assembly	1.LAP BELT ASSY & RESTRAINT SYSTEMS,AS LISTED IN PACIFIC SCIENTIFIC SB1108435-25-01/1108460-25-01.	N/A	-	1108460-25-01	Not applicable to P/N installed.
34	FAA 95-01-01	06-Feb-1995	To prevent failure of the transponder to respond properly to Mode S interrogations from Mode S ground stations and Traffic Collision Avoidance System (TCAS) II	Terra Corporation TRT 250 Series Transponders	N/A	-	SB-104 R1	Not applicable to P/N installed.
34	FAA 95-04-01	08-Mar-1995	To prevent significant delays in the Honeywell Standard Windshear Detection Systems (WSS) detecting hazardous windshear	equipped with Honeywell Standard Windshear Detection Systems (WSS)	N/A	-	-	Not applicable to P/N installed.
78	FAA 95-10-11	31-May-1995	To prevent a loss of the center drive units (CDU) brake holding feature, which could result in possible movement of the fan reverser translating cowl towards the deploy position in flight	1.CF6-80C2 series turbofan engines installed on, but not limited to, Airbus A300 and A310 series, Boeing 747 and 767 series	N/A	-	78-1002 R1	Not applicable to P/N installed.
25	FAA 95-18-03	10-Oct-1995	To prevent tube puncture of the ramp/slide evacuation system, which could delay or impede the evacuation of passengers during an emergency	Model 767 series airplanes, equipped with BFGoodrich off-wing ramp/slide evacuation systems having part number (P/N) 101630, 101655, or 101656	N/A	-	767-25-0218	Not applicable to P/N installed.
28	FAA 95-15-06	23-Aug-1995	TO PREVENT IMPROPER FUNCTIONING OF CERTAIN ACTUATORS,WHICH COULD RESULT IN A FUEL IMBALANCE DUE TO THE INABILITY OF THE FLIGHTCREW TO	MODEL 727,737 SERIES A/C,EQUIPPED WITH J.C.CARTER COMPANY FUEL VALVE	N/A	-	61163-28-08 R0	Not applicable to P/N installed.
33	FAA 95-22-01	04-Dec-1995	To prevent smoke, fire, electrical shock, and possible electromagnetic interference caused by high voltage arcing in the cabin which, if undetected, could result in personal hazard or loss of the aircraft	AEROSPACE LIGHTING CORPORATION(ALC) LAMP CONNECTORS,P/N 31.85.1.A;SERIES 66 FLUORESCENT LAMPS;POWER UNITS,P/N's TR-991,TR-992,AL-0546,AND AL-0514;AND POWER SUPPLIES;P/N's 1895D	N/A	-	IB 90-001	Not applicable to P/N installed.
34	FAA 95-26-15	26-Dec-1995	To ensure that the flightcrew is advised of the potential hazard associated with failure of the audio output of the CAS-81 TCAS	1.CAS-81 Traffic Alert and Collision Avoidance Systems (TCAS) installed in transport category airplanes 2.AAR:CAS-81 TCAS PROCESSOR (ALLIEDSIGNAL社 제품)	N/A		-	Not applicable to P/N installed.

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	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			
34	FAA 96-02-06	26-Feb-1996	To prevent significant delays in the Honeywell Standard Windshear Detection Systems (WSS) detecting hazardous windshear, which could lead to the loss of flight path control	EQUIPPED WITH HONEYWELL STANDARD WINDSHEAR DETECTION SYSTEMS(WSS)	N/A	-	-	Not applicable to P/N installed.
25	FAA 95-25-08	20-May-1996	To prevent possible explosion of water heaters that could cause personal injury and aircraft damage	ROYAL INVENTUM COMPANY DR1 & DR6 SERIES GALLEY WATER HEATERS.	N/A	-	DR1/DR6-25-4 R1	Not applicable to P/N installed.
25	FAA 96-12-09	28-Jul-1996	To prevent injury to hands during operation of the PTC Model 950 series passenger seats equipped with footrest assembly,	PTC Seating Products Division, B/E Aerospace, Model 950 Series Passenger Seats Equipped With Footrest Assembly	N/A	-	25-1192 R1	Not applicable to P/N installed.
35	FAA 97-11-10	23-Jun-1997	To prevent restricted oxygen flow, which, if not corrected, could cause serious injury to a passenger in need of emergency or first aid oxygen during flight	Puritan Bennett Aero Systems Company Series 174290 Constant Flow Airline Portable Oxygen Masks, Part Numbers 174290-14, 174290-24, 174290-34, 174290-44, and 174290-	N/A	-	174290-35-1	Not applicable to P/N installed.
35	FAA 97-18-03	22-Sep-1997	To prevent failure of the ultrasonic weld on the cone and seal assembly of the oxygen mask with consequent reduced oxygen flow through the mask, which could result in the crew not being able to obtain oxygen in an emergency situation	Puritan-Bennett Aero Systems Co., Cone and Seal Assemblies, part numbers 210543 and 210543-01	N/A	-	3500-97-14	Not applicable to P/N installed.
26	FAA 98-24-27	28-Jan-1999	To prevent fire extinguisher failure due to leakage, which could result in an uncontained fire and damage to the aircraft	First Technology Fire and Safety Ltd. Toilet Compartment Fire Extinguishers	N/A	-	26-110 R1	Not applicable to P/N installed.
35	FAA 99-08-21	02-Jun-1999	To prevent reduced oxygen consumption when passengers are required to use defective oxygen masks, which could result in passenger injury	Puritan-Bennett Aero Systems Company C351-2000 Series Passenger Oxygen Masks and Portable Oxygen Masks	N/A	-	C351-2000-35-1 R2	Not applicable to P/N installed.
23	FAA 99-12-06	23-Jun-1999	To prevent VHF navigation receiver interference from frequency modulation (FM) radio station broadcast frequencies, which could cause distortion of the navigation audio and deflection of the desired flight path of the airplane during landing operations with possible loss of control of the airplane	AlliedSignal Inc. VN 411B Very High Frequency (VHF) Navigation Receivers	N/A	-	VN411B-21	Not applicable to P/N installed.
34	FAA 99-23-24	06-Dec-1999	To ensure the ILS receiver provides the flight crew with accurate glideslope data	AlliedSignal, Instrument Landing System Navigation Receivers, as Installed in, but Not Limited to, Airbus Model A300 Series Airplanes and Boeing Model 747-100, -100B, -100B SUD, -200B, -200F, -200C, -300, 747SR, and 747SP Series Airplanes	N/A	-	RIA-32A-34-471-48	Not applicable to P/N installed.
34	FAA 99-23-22	29-Nov-1999	MODE "C" TRANSPONDER WITH SINGLE GILLHA	Various Transport Category Airplanes Equipped with Mode "C" Transponder(s) with Single Gillham Code Altitude Input	N/A	-	-	Not applicable to P/N installed.
22	FAA 99-24-10	14-Jan-2000	To detect and correct problems with the standby vacuum system before failure or malfunction and to provide operating procedures for the pilot regarding the use and limitations of this system	Precise Flight, Inc. Model SVS III Standby Vacuum Systems	N/A	-	-	Not applicable to P/N installed.
34	FAA 99-23-22R1	29-Nov-1999	REPETITIVE TEST TO DETECT DISCREPANCIES	Various Transport Category Airplanes Equipped with Mode "C" Transponder(s) with Single Gillham Code Altitude Input	N/A	-	-	Not applicable to P/N installed.

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7. HL7742 Appliance AD

As of 23.Jul.2013

ATA	FAA AD		AD DESCRIPTION	EFFECTIVITY		COMPLIANCE	RELATED DOCUMENT	HL7742 Appliance
	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			
34	FAA 2000-07-27	03-May-2000	To prevent loss of the main sources of attitude data, consequent high pilot workload, and a significant increase in the likelihood of pilot error	Various Transport Category Airplanes Equipped With Certain Honeywell Air Data Inertial Reference Units	N/A	-	HG2030AD-34-0008	Not applicable to P/N installed.
0	FAA 2000-04-12	15-Feb-2000	To prevent titanium propane cylinders from cracking and releasing propane gas vapor while the balloon is in service	Cameron Balloons Ltd. (Thunder & Colt) Titanium Propane Cylinders, part number (P/N) CB2380 and P/N CB2383	N/A	-	-	Not applicable to P/N installed.
34	FAA 94-01-06	04-Feb-1994	COLLINS-TCAS II COMPUTER-INST. NEW TCAS	1.B737,B767,B747 EQUIPPED WITH COLLINS TCAS II COMPUTER (P/N 622-8971-012) 2.AAR : 7413/7413/7415/7416/7417/7418	N/A	-	TTR-920-34-16	Not applicable to P/N installed.
34	FAA 2002-06-06	03-May-2002	To prevent erroneous altitude resolutions from causing a reduction in the intended aircraft collision avoidance system (ACAS) or traffic alert and collision avoidance system (TCAS) Change 7 minimum separation	Rockwell Collins, Inc. TDR-94 and TDR-94D Mode S Transponders	N/A	-	TDR-94/94D-34-17	Not applicable to P/N installed.
34	FAA 2002-06-05	26-Apr-2002	To prevent transmission of inaccurate data concerning altitude from one airplane to another, which could cause the pilot receiving the data to change course, either ascending or descending, and possibly lead to a mid-air collision or near mid-air collision	Various Transport Category Airplanes Equipped With Air Traffic Control (ATC) Transponders Manufactured by Rockwell Collins, Inc	N/A	-	621A-3-34-21	Not applicable to P/N installed.
34	FAA 2002-08-09	09-Aug-2002	To ensure the correct transition of the IRU to battery power upon the loss of primary power. Failure of an IRU to transition to backup battery power could result in loss of attitude, heading, and position reference and lead to the pilot making flight decisions that put the aircraft in unsafe flight conditions	Honeywell, Inc. Part Number HG1075AB05 and HG1075GB05 Inertial Reference Units	N/A	-	HG1075AB-34-A0013	Not applicable to P/N installed.
34	FAA 2002-14-19	23-Aug-2002	To prevent an unwarranted display of the ADC flag when switching static air sources	Rockwell Collins, Inc. ADC-85, ADC-85A, ADC-850D, and ADC-850F Air Data Computers	N/A	-	-	Not applicable to P/N installed.
25	FAA 2002-21-01	27-Nov-2002	INSPECTION THE WIRING FOR INDICATION OF OVERHEATING OR ELECTRICAL ARCING	BRITAX SELL GmbH & Co.OHG WATER BOILER, COFFEE MAKERS & BEVERAGE MAKERS	N/A	-	E33-4-007SB R2 E33-4-009SB R0 E33-4-011SB R2 E33-4-012SB R1 E33-4-013SB R0 E33-4-015SB R1 E33-4-010SB R1 E33-4-014SB R1 E33-4-016SB R1	Not applicable to P/N installed.
31	FAA 2002-20-09	12-Nov-2002	To prevent premature failure of the Application Specific Integrated Circuit (ASIC)	Rockwell Collins, Inc. AFD-3010 Adaptive Flight Display Units	N/A	-	AFD-3010-31-12 R2	Not applicable to P/N installed.
25	FAA 2002-21-01 SUPERSEDED AD 2001-10-13	27-Nov-2002	To prevent a fire in the galley compartment due to inadequate crimping of the electrical terminal contact pins, which could result in smoke in the cockpit and cabin and loss of control of the airplane	Britax Sell GmbH & Co. OHG Water Boilers, Coffee Makers, and Beverage Makers	N/A	-	E33-4-007SB R2 E33-4-009SB R0 E33-4-011SB R2 E33-4-012SB R1 E33-4-013SB R0 E33-4-015SB R1 E33-4-010SB R1 E33-4-014SB R1 E33-4-016SB R1	Not applicable to P/N installed.
73	FAA 2002-22-12	13-Dec-2002	To prevent failure of a hose when exposed to fire	Titeflex Corporation high-pressure and medium-pressure hoses installed on Airbus A300, A310, A340 airplanes, Boeing airplane models 707, 727, 737-200, 737-200C, 747, 757,	N/A	-	73-2	Not applicable to P/N installed.

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ATA	FAA AD		AD DESCRIPTION	EFFECTIVITY		COMPLIANCE	RELATED DOCUMENT	HL7742 Appliance
	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			
34	FAA 2002-22-13	20-Dec-2002	to prevent the FMC from retaining original information when an edit is made to a procedure or flight plan. Such a condition could cause the pilot to fly the airplane out of the range of the correct altitude constraint	Rockwell Collins, Inc. FMC-4200, FMC- 5000, and FMC-6000 Flight Management Computers	N/A	-	-	Not applicable to P/N installed.
34	FAA 2003-04-06	11-Mar-2003	To ensure that the flightcrew has an accurate glideslope deviation indication	HONEYWELL PRIMUS II RNZ-850/-851 INU	N/A	-	-	Not applicable to P/N installed.
25	FAA 2003-03-11	2003-03-07	1.One-time unpacking and subsequent repacking of the slide/raft system 2.Repacking of all other slide/raft systems at the next required normal maintenance schedule of the slide/raft system	Air Cruisers co. Emergency Evacuation Slide/Raft system P/N 62774-401~408	N/A	-	777-107-25-06	Not applicable to P/N installed.
34	FAA 2003-13-08	21-Jul-2003	To prevent the loading of the baro set potentiometer	Aircraft equipped with TAWS Model TAWS8000 Manufactured by Goodrich Avionic System	N/A	-	Goodrich Avionics Service Memo 134	Not applicable to P/N installed.
25	FAA 2003-26-06	17-Feb-2004	INSP' SAFETY BELTS & RESTRAINT SYS' FOR DEFECTS & SVC LIFE LIMITS	ANJOU Aeronautique Co. SAFETY BELTS & RESTRAIN SYSTEM (P/N 343, 343-1, 343AM, 343C, 343BM, 343CM, 343D, 343M)	N/A	-	-	Not applicable to P/N installed.
26	FAA 2003-26-14	20-Feb-2004	REMOVE THE AFFECTED FIRE EXTINGUISHER FROM SVC & WOULD PREVENT YOU FROM USING THEM.	HAND-HELD FIRE EXTINGUISHERS P/N: 898052 S/N : V-432001 ~ W389653	N/A	-	898052-26-449 R0	Not applicable to P/N installed.
25	FAA 2004-03-01	11-Mar-2004	To prevent failure of the slide/raft to properly inflate.	All dash numbers of Air Cruisers Company Emergency Evacuation Slide/Raft System, P/N 62774. These Emergency Evacuation Slide/Raft Systems are installed on, but not limited to Boeing B777-200 and -300 series airplanes.	N/A	-	SB 777-107-25-06 REV.G	Not applicable to P/N installed.
34	FAA 2004-08-15	07-Jun-2004	To prevent the loading of the baro set potentiometer	All airplane models and serial numbers, certificated in any category, that incprporate a Goodrich TAWS8000 terrain awareness warning system (TAWS), P/N 805-18000-001, with "Mod None", "Mod A", or "Mod B" hardware installed.	N/A	-	Goodrich Avionics Service Memo 134	Not applicable to P/N installed.
34	FAA 2004-08-16	01-Jun-2004	Modification to the transponder by adding a resistor and transistor to the circuit board	NARCO Avionics Inc. AT150 transponders with "Chassis Level A", serial number (S/Ns) 10000 through 12598 inclusive	N/A	-	SB AT150 NO.1	Not applicable to P/N installed.
34	FAA 2004-10-15	09-Jul-2004	GARMIN International Inc. GTX 330/ GTX 330D Mode S Transponders Upgrade Version 3.03, 3.04, or 3.05	GARMIN International Inc. GTX330/330D Mode S transponders intalled on airplane	N/A	-	NONE	Not applicable to P/N installed.
34	FAA 2004-13-20	03-Aug-2004	MOD' & TEST OF SW FOR APOLLO GX50 ETC GPS NAV' UNIT	Apollo GX Series GPS Navigation unit Apollo GX50/55/60/65	N/A	-	561-4002-001 R0	Not applicable to P/N installed.

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ATA	FAA AD		AD DESCRIPTION	EFFECTIVITY		COMPLIANCE	RELATED DOCUMENT	HL7742 Appliance
	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			
70	FAA 2004-21-05	19-Nov-2004	To prevent combustion by products (carbon-monoxide exhaust) and fuel leakage from the combustion heaters caused by failure of the combustion heater system	Kelly Aerospace Power Systems B-Series Combustion Heaters, Models B1500,B2030,B2500,B3040,B3500,B4050,and B4500, marked as meeting the standards of TSO-C20, that are installed on, but not limited to, the following aircraft (all serial numbers), and are certificated in any category.	N/A	-	#A-103	Not applicable to P/N installed.
25	FAA 2004-22-01	01-Dec-2004	To prevent shifting or unrestrained cargo in the cargo compartment.	The Transport category airplanes, certificated in any category, on which cargo restraint strap assemblies P/N 1519-MCIDS have been installed per Supplemental Type Certificate (STC) ST01004NY	N/A	-	STC ST01004NY	Not applicable to P/N installed.
34	FAA 2005-01-19	23-Feb-2004	To prevent interrogating aircraft from possibly receiving inaccurate replies	GARMIN International Inc. GTX330/330D Mode S transponders intalled on airplane	N/A	-	-	Not applicable to P/N installed.
26	FAA 2006-04-04	24-Mar-2006	To Identify And Provide Corrective Action For A Potentially Inoperative Smoke etector (Applicable to Meggitt Model 602 Smoke Detectors)	Meggitt Smoke Detector Model 602 (P/N 8930-())	N/A	-	Meggitt SIL 8930-26-01 Rev.C FAA AD 2005-02-04	Not applicable to P/N installed.
25	FAA 2006-12-08	17-Jul-2006	To prevent loss of pressure in the escape slides/rafts after an emergency evacuation	Goodrich Evacuation Systems Approved Under Technical Standard Order (TSO) TSO-C69b, as installed on Airbus Model A330-201, -202, -203, -223, -243, -301, -321, -322, -323, -341, -342, and -343 airplanes; Model A340-211, -212, -213, -311, -312, and -313 airplanes; and Model A340-541 and -642 airplanes; certificated in any category	HL7736/40/ Spare Part (Slide/Raft 7A1508-119/7A1509-119/7A-1539-119/7A1539-120 of SB 25-355 Group1	-	Goodrich SB 25-355 R1	Not applicable to P/N installed.
25	FAA 2007-23-01	11-Dec-2007	To prevent loss of pressure in the escape slides/rafts after an emergency evacuation	Goodrich evacuation systems approved under Technical Standard Order (TSO) TSO-C69b, as installed on Airbus Model A330-201, -202, -203, -223, -243, -301, -321, -322, -323, -341, -342, and -343 airplanes; Model A340-211, -212, -213, -311, -312, and -313 airplanes; and Model A340-541 and -642 airplanes	A330 Part	-	Goodrich SB 25-355 R1	Not applicable to P/N installed.
25	FAA 2008-03-05	11-Mar-2008	To prevent failure of an evacuation system	Boeing Model 747-100, 747-100B, 747-100B SUD, 747-200B, 747- 200C, 747-200F, 747-300, 747-400, 747-400D, 747-400F, 747SR, and 747SP series airplanes; and Model 767-200 and -300 series airplanes; certificated in any category; equipped with any Goodrich evacuation system listed in Table 1 of this AD	All B747-400 and B767-300 equipped with any Goodrich evacuation system	-	Goodrich SB 25-343 R3	Not applicable to P/N installed.

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ATA	FAA AD		AD DESCRIPTION	EFFECTIVITY		COMPLIANCE	RELATED DOCUMENT	HL7742 Appliance
	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			
25	FAA 2008-06-27	28-Apr-2008	To prevent failure of an evacuation system	(1) Goodrich evacuation systems approved under Technical Standard Orders (TSOs) TSO-C69, TSO-C69a, and TSO-C69b, installed on certain Boeing airplanes, certificated in any category, as listed in Table 1 of this AD. (2) Goodrich evacuation systems approved under TSOs TSO-C69, TSO-C69a, and TSO-C69b, installed on certain Airbus airplanes, certificated in any category, as listed in Table 3 of this AD.	B737/B747/B767	-	Goodrich SB 25-343 R3	Not applicable to P/N installed.
34	FAA 2008-06-28 R1	10-Apr-2008	Incorrect Information Display on PFD (Primary Flight Display)	Certain A/C equipped with Avidyne PFD (P/N 700-00006-000/001/002/003/100)	N/A	-	Avidyne Service Alert No. SA-08-001	Not applicable to P/N installed.
34	FAA 2009-05-05	12-Mar-2009	To prevent certain conditions from existing when PFDs display incorrect attitude, altitude, and airspeed information	Avidyne Corporation (Avidyne) Primary Flight Displays (PFDs), part numbers (P/Ns) 700-00006-000, 700-00006-001, 700-00006-002, 700-00006-003, and 700-00006- 100 with any serial number listed in Avidyne Service Bulletin No. 601-00006-096, Revision 1,	N/A	-	Avidyne SB 601-00006-096 R1	Not applicable to P/N installed.
35	FAA 2009-21-10	17-Dec-2009	To prevent an oxygen cylinder from rupturing, which, depending on the location, could result in structural damage and rapid decompression of the airplane,	Crew Oxygen Cylinder manufactured by AVOX System & B/E Aerospace, S/N STXXXXX & KXXXXXX	N/A	-	SB 737-35A1124 SB 767-35A0058 SB 747-35A2127 SB 777-35A0023 AOT A320-35A1042 AOT A330-35A3025 VSB 6084-34-35-01 MWD ME-09-123	Not applicable to P/N installed.
26	FAA 2010-01-03	20-Jan-2010	To replace portable Halon 1211 (BCF) fire extinguishers manufactured by Fire Fighting Enterprises Limited with serviceable fire extinguishers	portable Halon 1211 (BCF) fire extinguishers manufactured by Fire Fighting Enterprises Limited	N/A	-	EASA AD 2009-0251-E EASA AD 2009-0262	Not applicable to P/N installed.
35	FAA 2009-21-10 R1	04-Feb-2010	To prevent an oxygen cylinder from rupturing, which, depending on the location, could result in structural damage and rapid decompression of the airplane,	Crew Oxygen Cylinder manufactured by AVOX System & B/E Aerospace, S/N STXXXXX & KXXXXXX	N/A	-	SB 737-35A1124 SB 767-35A0058 SB 747-35A2127 SB 777-35A0023 AOT A320-35A1042 AOT A330-35A3025 VSB 6084-34-35-01 MWD ME-09-123	Not applicable to P/N installed.
25	FAA 2001-10-13	15-Jun-2001	INSPECTION OF WIRES ON THE TEMPERATURE LIMITER INSTALLED ON COFFEE MAKER, WATER BOILERS	Britax Sell GmbH & Co. OHG water boilers, coffee makers, and beverage makers, listed by part number (P/N) and serial number (SN) in Table 1 of this AD,installed on, but not limited to, Airbus Industrie A319, A320, A330, AVRO RJ, Bombardier DHC-8-400, and Boeing Company 717, 737, 747, 757, 767, 777	N/A	-	SB E33-4-010SB SB E33-4-011SB	Not applicable to P/N installed.

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ATA	FAA AD		AD DESCRIPTION	EFFECTIVITY		COMPLIANCE	RELATED DOCUMENT	HL7742 Appliance
	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			
26	FAA 2003-23-05	2003-23-05	Titeflex Hoses/B-nuts; inspection of hoses for proper date and paint code and inspection of aluminum B-nuts for proper heat treatment	BACH5R0080YY BACH5R0140YU BACH5S0140XT BACH5R0186YY BACH5R0186XX BACH5S0080XX BACH5S0080YY BACH5S0110XN	N/A	-	SB 747-26A2269	N/A
35	FAA 2010-11-05	06-Jul-2010	To prevent an a high-pressure gaseous oxygen cylinder from rupturing.	Oxygen Cylinder listed in SB 6084-34-35-01 R1	AVOX Portable Oxygen Cylinders	1. Within 60days from effective date : Embodiment of SB 6084-34-35-01 2. From effective date : AD compliance	SB 6084-34-35-01 R0/R1	Completed SB 777-35A0023 2010.01.13
25	FAA 2011-01-09	09-Feb-2011	To prevent PBE units from igniting, which could result in a fire and possible injury to the flightcrew or other persons.	B/E Aerospace protective breathing equipment (PBE) units having part number (P/N) 119003-11 S/N From 003-50730M to 003-51329M	N/A	-	SB 119003-35-5 SB 119003-35-6	Not applicable to P/N installed.
35	FAA 2011-04-09 (Security)	11-Feb-2011	Removing all chemical oxygen generators in the lavatories, or activating all oxygen generators until the generator oxygen supply is expended	Operating under 14 CFR part 121/U.S.-registered and operating under 14 CFR part 129, with a maximum passenger capacity of 20 or greater.	AAR ALL FLEET EXCEPT CARGO	Within Apr 04, 2011	None (PER SSAD)	Completed MEO CB CB1-25-0007 2011.02.23 SS
35	FAA 2011-14-08	19-Aug-2011	To prevent the inline flow indicators of the oxygen mask assembly from fracturing and separating	B/E Aerospace, Continuous Flow Passenger Oxygen Mask Assembly Part Numbers 174006-( ), 174080-( ), 174085-( ), 174095-( ), 174097-( ), and 174098-( ) as listed in B/E Aerospace Service Bulletin 174080-35-04, Rev	HL7596/97/55/56/7775/7791	-	SB 174080-35-04 R0 Ref) SB 174080-35-02	N/A by Part
25	FAA 2011-21-08	23-Nov-2011	General visual inspection for cracking of the backrest links/ replacement with new.	Sicma Aero Seat 88xx, 89xx, 90xx, 91xx, 92xx, 93xx, 95xx, and 96xx series passenger seat assemblies	HL7596/97/55/56/7775/7791	N/A	SB 90-25-013 DGAC 2001-613(AB)	Not applicable to P/N installed.
34	FAA 2012-02-08	13-Mar-2012	To prevent TCAS units from dropping tracks, which could compromise separation of air traffic and lead to subsequent mid-air collisions.	ACSS TCAS with part numbers identified in ACSS Technical Newsletter 8008359 (HL7620 Applicable)	HL7620	Within Mar 12, 2015	Technical Newsletter 8008359 Rev.B AMOC SB 8008559-001 (ACSS SB 7517900-34-6048)	Not applicable to P/N installed.
25	FAA 2012-06-25	04-May-2012	To prevent loss of pressure in the escape slides/rafts after an emergency evacuation	Goodrich evacuation systems approved under Technical Standard Order (TSO) TSO-C69b, as installed on Airbus Model A330-201, -202, -203, -223, -243, -301, -321, -322, -323, -341, -342, and -343 airplanes; Model A340-211, -212, -213, -311, -312, and -313 airplanes; and Model A340-541 and -642 airplanes	AAR A330 Escape Slide/Rafts as identified SB 7A1508/09/10/39-25-373 R3.	Within May 03, 2015	Goodrich SB 7A1508/09/10/39-25-373 R3	Not applicable to P/N installed.
35	FAA 2012-11-09	10-Aug-2012	To eliminate this hazard and ensure that all lavatories have a supplemental oxygen supply.	(1) Airplanes that are in compliance with the requirements of AD 2011-04-09, Amendment 39-16630 (76 FR 12556, March 8, 2011). (2) Airplanes equipped with any chemical oxygen generator installed in any lavatory and are: (i) Operating under 14 CFR part 121; or (ii) U.S.-registered and operating under 14 CFR part 129, with a maximum passenger capacity of 20 or greater	AAR ALL FLEET EXCEPT B747 Airplane & B767 CARGO	Within 2015.09.09	FAA 2011-04-09	Open

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ATA	FAA AD		AD DESCRIPTION	EFFECTIVITY		COMPLIANCE	RELATED DOCUMENT	HL7742 Appliance
	AD No.	EFFECTIVE DATE (DD-MMM-YYYY)		AD	AAR			
34	FAA 2012-26-15	24-Jan-2013	To detect and correct inaccuracies of the pressure sensors, which could result in altitude,computed airspeed, true airspeed, and Mach computation errors.	The part numbers and serial numbers identified in Honeywell SB ADM/ADC/ADAHRS-34-A01, dated November 6, 2012	A321(HL8256) having P/N PG1152BC02, S/N 1202W877/1202W906(2EA)	Within 30 days from the effective date of this AD. (Within 2013.01.24)	SB ADM/ADC/ADAHRS-34-A01	N/A by Part

8-1. Airframe SB Status

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-20-0001	SB 777-20-0001	000	00	STANDARD PRACTICES - ELECTRICAL/ELETRONICS SAFETRY EQUIPMENT - WIRE INSPECTION AND CLAMP CONFIGURAATION CHANGE	COMPLETED	2012.12.02	N/A
SB 777-21-0106	SB 777-21-0106	000	00	PACK COOLING AND MIX MANIFOLD TEMPERATURE CONTROL - L/H Fan Inlet Diffuser Housing (FIDH) - Replacement.	COMPLETED	2012.09.18	N/A
SB 777-21-0106	SB 777-21-0106	001	00	PACK COOLING AND MIX MANIFOLD TEMPERATURE CONTROL - R/H Fan Inlet Diffuser Housing (FIDH) - Replacement.	COMPLETED	2012.12.06	N/A
SB 777-21-0114	SB 777-21-0114	000	00	B777 ELMS2 Panel 100/200/300 Replacement	COMPLETED	2010.04.19	N/A
SB 777-21-0142	SB 777-21-0142	000	02	AIR CONDITIONING - Pack Cooling and Mix Manifold Temperature Control - Installation of Ram Air Exit Louvers	COMPLETED	2012.12.10	N/A
SB 777-21-0146	SB 777-21-0146	000	02	AIR CONDITIONING - L/H Ram Air Inlet Door and Actuator - Inspection and Replacement of Ram Air Inlet Door Control Linkage	COMPLETED	2012.12.07	N/A
SB 777-21-0146	SB 777-21-0146	001	02	AIR CONDITIONING - R/H Ram Air Inlet Door and Actuator - Inspection and Replacement of Ram Air Inlet Door Control Linkage	COMPLETED	2012.12.07	N/A
SB 777-22-0017	SB 777-22-0017	000	00	AUTOFLIGHT - Autopilot Flight Director System - Autopilot Flight Director Computer Software Change	COMPLETED	2007.08.30	N/A
SB 777-22A0024	SB 777-22A0024	000	00	AUTOFLIGHT - Autopilot Flight Director System - Install Autopilot Flight Director Computer Operational Program Software	COMPLETED	2010.02.16	N/A
SB 777-23-0238	SB 777-23-0238	000	00	COMMUNICATIONS - Passenger Service System - Drip Shield Installation on Zone Management Unit	COMPLETED	2010.05.06	N/A

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-23-0246	SB 777-23-0246	001	01	COMMUNICATIONS - Cabin Management System (CMS) - Passenger Address / Cabin Interphone (PA/CI) Operational Software (OPS) and Overhead Electronics Units(OEU) OPS Update.	COMPLETED	2010.12.27	N/A
SB 777-23-0295	SB 777-23-0295	000	01	COMM - PSS - OPS(Operational S/W) Change	OPEN		N/A
SB 777-23-0328	SB 777-23-0328	000	00	Passenger Entertainment System Video - Wire Routing Change to Remote Cabin Management Terminal	OPEN		N/A
SB 777-24-0099	SB 777-24-0099	000	02	Electrical Load Management (ELM) System - Power Panel - Monitor Card Inspection and Replacement.	OPEN		N/A
SB 777-24-0101	SB 777-24-0101	000	02	ELECTRICIAL POWER - Connexion by Boeing (CbB) - System Removal.	COMPLETED	2010.04.19	N/A
SB 777-24-0105	SB 777-24-0105	000	00	ELECTRICAL POWER - Power and Regulation - Electrical Load Management System Terminal Lug Installation Change.	COMPLETED	2008.01.20	N/A
SB 777-24-0106	SB 777-24-0106	000	00	ELECTRICAL POWER - Electrical Load Management (ELM) System - Installation of Enclosure Trays for Debris Containment.	COMPLETED	2010.04.14	N/A
SB 777-24-0109	SB 777-24-0109	001	00	ELECTRICAL POWER - Batteries - Main/APU Battery Charger Replacement	COMPLETED	2011.01.17	N/A
SB 777-24-0109	SB 777-24-0109	002	00	ELECTRICAL POWER - Batteries - Main/APU Battery Charger Replacement	COMPLETED	2011.01.24	N/A
SB 777-24-0111	SB 777-24-0111	000	00	ELECTRICAL POWER - Electrical Load Management (ELM) System - Power Panel Inspection	COMPLETED	2010.04.14	N/A
SB 777-24-0112	SB 777-24-0112	000	02	ELM System - P200 and P300 Power Panels - Contactor Replacement	OPEN		N/A

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-24-0113	SB 777-24-0113	000	00	AC Generation And Bus Control - Generator Control Unit Software Change	COMPLETED	2013.05.18	N/A
SB 777-24-0117	SB 777-24-0117	000	00	AC External Power - Bus Power Control Unit Software Change	COMPLETED	2013.05.18	N/A
SB 777-24-0121	SB 777-24-0121	000	00	ELECTRICAL POWER - Backup Electrical Power - Backup Generator Converter Software Installation	COMPLETED	2012.12.06	N/A
SB 777-24-0123	SB 777-24-0123	000	00	External Power Receptacle - Foreign Object Debris Shield Change	OPEN		N/A
SB 777-24A0119	SB 777-24A0119	000	00	Power and Regulation - Inspection and if Necessary, Repair of Clamps, Power Feeder Cable, Insulation Blanket and Aft Pressure Bulkhead	COMPLETED	2012.12.05	N/A
SB 777-25-0427	SB 777-25-0427	000	00	EQUIPMENT / FURNISHINGS - Passenger compartment Lining - Door 1 Left Side and Door 1 Right Side Safety Strap Bracket Assembly Change.	COMPLETED	2012.12.02	N/A
SB 777-26-0045	SB 777-26-0045	000	00	FIRE PROTECTION - Lower Cargo Fire Extinguishing - Filter/Regulator Change	COMPLETED	2009.07.25	N/A
SB 777-26-0056	SB 777-26-0056	000	00	Cargo Compartment Smoke Detection - Water Separator Tubing System Inspection	OPEN		N/A
SB 777-27-0062	SB 777-27-0062	000	00	Elevator and Rudder Control System - Elevator, Rudder and Rudder Tab - Freeplay Inspection and Lubrication	COMPLETED(Initial Insp`)	2007.08.30	FAA 2007-13-05_000_00
SB 777-27-0062	SB 777-27-0062	001	00	Freeplay Check for the Elevator	COMPLETED(Initial Insp`)	2008.11.27	FAA 2007-13-05_001_00
SB 777-27-0062	SB 777-27-0062	003	00	Freeplay Check for the Rudder Tab	COMPLETED(Initial Insp`)	2008.11.27	FAA 2007-13-05_003_00

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-27-0062	SB 777-27-0062	002	00	Freeplay Check for the Rudder	COMPLETED(Initial Insp`)	2008.11.28	FAA 2007-13-05_002_00
SB 777-27-0069	SB 777-27-0069	000	01	FLIGHT CONTROLS - Primary Flight Computer Software - Revision	COMPLETED	2007.08.31	N/A
SB 777-27-0072	SB 777-27-0072	000	00	Aileron and Flaperon Control - Wheel Force Transducer - Wire Routing Revision.	COMPLETED	2010.04.15	N/A
SB 777-27-0079	SB 777-27-0079	001	01	Primary Flight Control System - Primary Flight Computer Software Change	COMPLETED	2013.03.12	N/A
SB 777-27-0080	SB 777-27-0080	000	00	FLIGHT CONTROLS - Primary Flight Control	COMPLETED	2009.12.14	N/A
SB 777-27-0100	SB 777-27-0100	000	01	FLIGHT CONTROLS - Primary Flight Compute	COMPLETED	2012.12.09	N/A
SB 777-27-0104	SB 777-27-0104	000	00	Inspection and Replacement Of Fasteners	COMPLETED(Initial Insp`)	2011.05.20	N/A
SB 777-27-0106	SB 777-27-0106	000	00	Stall Warning - Switch Guard Installatio	COMPLETED	2011.02.02	N/A
SB 777-27-0107	SB 777-27-0107	000	00	Spoiler and Speedbrake Control System -	OPEN		N/A
SB 777-27A0059	SB 777-27A0059	000	00	Hori' Stab' DVI, Lub' & Freeplay Check	COMPLETED(Initial Insp`)	2008.11.28	N/A
SB 777-27A0078	SB 777-27A0078	00	00	LEFT and RIGHT HORIZONTAL STABILIZER BON	COMPLETED(Initial Insp`)	2012.12.09	N/A

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-27A0078	SB 777-27A0078	001	00	HORIZONTAL STABILIZER TO BODY BONDING JU	COMPLETED(Initial Insp`)	2012.12.03	N/A
SB 777-27A0078	SB 777-27A0078	002	00	VERTICAL FIN TIP AREA BONDING JUMPER INS	COMPLETED(Initial Insp`)	2012.12.03	N/A
SB 777-27A0078	SB 777-27A0078	003	00	VERTICAL FIN Tip and BODY BONDING JUMPER	COMPLETED(Initial Insp`)	2012.12.03	N/A
SB 777-27A0078	SB 777-27A0078	004	00	Left Aileron Bonding Jumper Inspection,	COMPLETED(Initial Insp`)	2012.12.04	N/A
SB 777-27A0078	SB 777-27A0078	005	00	Right Aileron Bonding Jumper Inspection,	COMPLETED(Initial Insp`)	2012.12.04	N/A
SB 777-27A0078	SB 777-27A0078	000	01	Vertical Fin Tip Area - SUPPLEMENTAL INS	OPEN (Terminating Action, If Abnoraml Condition Found During Repeat Insp`)		N/A
SB 777-27A0078	SB 777-27A0078	006	01	Left Aileron Bonding Jumper Replacement	OPEN(Terminating Action, If Abnoraml Condition Found During Repeat Insp`)		7A-27-AAR-1F-01_000_00
SB 777-27A0078	SB 777-27A0078	007	01	Right Aileron Bonding Jumper Replacement	OPEN(Terminating Action, If Abnoraml Condition Found During Repeat Insp`)		7A-27-AAR-1F-02_000_00
SB 777-27A0078	SB 777-27A0078	008	01	LEFT and RIGHT HORIZONTAL STABILIZER BON	OPEN(Terminating Action, If Abnoraml Condition Found During Repeat Insp`)		7A-27-AAR-1F-03_000_00
SB 777-27A0078	SB 777-27A0078	009	01	HORIZONTAL STABILIZER TO BODY BONDING JU	OPEN(Terminating Action, If Abnoraml Condition Found During Repeat Insp`)		7A-27-AAR-1F-04_000_00
SB 777-27A0078	SB 777-27A0078	010	01	VERTICAL FIN TIP AREA BONDING JUMPER REP	OPEN(Terminating Action, If Abnoraml Condition Found During Repeat Insp`)		7A-27-AAR-1F-05_000_00

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-27A0078	SB 777-27A0078	011	01	VERTICAL FIN TO BODY BONDING JUMPER	OPEN(Terminating Action, If Abnoraml Condition Found During Repeat Insp`)		7A-27-AAR-1F-06_000_00
SB 777-28-0048	SB 777-28-0048	001	00	FUEL - Indicating - Fuel Quantity Processor Unit Software Change.	COMPLETED	2011.12.30	N/A
SB 777-28-0050	SB 777-28-0050	000	00	FUEL - Indicating - Fuel Quantity Processor Unit Software Change	COMPLETED	2009.02.08	N/A
SB 777-28-0061	SB 777-28-0061	000	01	FUEL - Distribution - Left and Right Engine Fuel Spar Valve Control Relay and Wiring Location Change.	OPEN		N/A
SB 777-28-0065	SB 777-28-0065	000	00	FUELS - Engine Fuel Feed System - Wire Bundle Routing Change.	COMPLETED	2012.12.04	N/A
SB 777-28-0066	SB 777-28-0066	000	00	FUEL - Indicating - Fuel Quantity Processor Unit Replacement	OPEN		N/A
SB 777-28-0067	SB 777-28-0067	000	00	FUEL - Indicating - Fuel Quantity Processor Unit Software Change	COMPLETED	2011.12.30	N/A
SB 777-28-0068	SB 777-28-0068	000	02	Engine Fuel Feed System - Crossfeed Valve Control Electrical Wiring Separation	OPEN		N/A
SB 777-28-0073	SB 777-28-0073	000	01	Work Package 1 - Wire Harness Cross Pinning Inspection	COMPLETED	2012.12.07	N/A
SB 777-28A0038	SB 777-28A0038	000	00	FUEL - Engine Fuel Feed System - Main Tank Boost Pump Ground Fault Protection Change	COMPLETED	2010.04.19	N/A
SB 777-28A0040	SB 777-28A0040	000	00	Fuel Jettison System - Fuel Pump Inlet Protection - Center Override/Jettison Fuel Pump - Electrical Load Managment System (ELMS) - Software Change.	COMPLETED	2010.04.14	N/A

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-28A0047	SB 777-28A0047	001	03	Fuel Jettison System - Center Override/Jettison Fuel Pumps and Main Jettison Fuel Pumps - Change to Give Redundant Control of the Pump	COMPLETED	2010.04.19	N/A
SB 777-29-0032	SB 777-29-0032	000	00	Fuel Pumps - Change to Give Redundant Control of the Pump	COMPLETED	2008.01.15	N/A
SB 777-29-0035	SB 777-29-0035	000	00	B777 Hyd Equipment Support Frame - Insp	COMPLETED	2010.04.07	N/A
SB 777-29-0035	SB 777-29-0035	001	00	B777 Hyd Equipment Support Frame - Insp	COMPLETED	2010.04.07	N/A
SB 777-29-0037	SB 777-29-0037	000	00	B777 CWT Hydraulic Tube Clamp Block Repl	COMPLETED	2011.05.20	N/A
SB 777-29-0038	SB 777-29-0038	000	01	B777 Hyd Brake Line Adapter Replacement	OPEN		N/A
SB 777-30-0021	SB 777-30-0021	000	00	L/H WING TAI DUCT CONNECTION SCREW CHANG	COMPLETED	2010.04.08	N/A
SB 777-30-0021	SB 777-30-0021	001	00	R/H WING TAI DUCT CONNECTION SCREW CHANG	COMPLETED	2010.04.08	N/A
SB 777-30-0022	SB 777-30-0022	000	00	Water and Drain Line Heaters - Drain Mas	COMPLETED	2010.04.15	N/A
SB 777-30-0023	SB 777-30-0023	000	00	Ice Detection System - Ice Detector Elec	OPEN		N/A
SB 777-31-0097	SB 777-31-0097	000	03	INDICATING AND RECORDING SYSTEMS - Airplane Information Management System - Blockpoint 2005A Software Upgrade (AIMS-2 only).	COMPLETED	2008.10.16	N/A

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-31-0108	SB 777-31-0108	000	00	Indicating/Recording Systems - Display Unit - Install Updated Phase 3 Airplane Information Management System Display Unit Database	COMPLETED	2009.10.30	N/A
SB 777-31-0152	SB 777-31-0152	000	00	Indicating/Recording Systems - Warning E	COMPLETED	2009.12.14	N/A
SB 777-31-0159	SB 777-31-0159	000	01	INDICATING/RECORDING SYSTEMS - Systems Card Files - ARINC Signal Gateway (ASG) Software Revision	COMPLETED	2010.08.23	N/A
SB 777-31-0167	SB 777-31-0167	000	00	INDICATING/RECORDING SYSTEMS - Airplane Information Management System - Blockpoint V15 Software Change.	COMPLETED	2011.05.20	N/A
SB 777-31-0187	SB 777-31-0187	000	00	INDICATING/RECORDING SYSTEMS - Primary Display System - Display Unit Replacement and Software Change	COMPLETED	2013.03.28	N/A
SB 777-31A0120	SB 777-31A0120	000	02	INDICATING AND RECORDING SYSTEMS - Airplane Information Management System - AIMS-2 Blockpoint 2006 Software Upgrade.	COMPLETED	2008.10.13	N/A
SB 777-31A0150	SB 777-31A0150	000	01	INDICATING/RECORDING SYSTEMS - Airplane Information Management System - AIMS Blockpoint V14 Software Upgrade (AIMS-2 Airplanes)	COMPLETED	2010.08.25	N/A
SB 777-32-0056	SB 777-32-0056	000	02	B777 LMG J501&502 Junction Box Replaceme	COMPLETED	2011.05.20	N/A
SB 777-32-0056	SB 777-32-0056	001	02	B777 RMG J501&502 Junction Box Replaceme	COMPLETED	2011.05.20	N/A
SB 777-32-0058	SB 777-32-0058	000	01	B777 MLG Upper & Lower Hyd Support Repl	COMPLETED	2010.04.17	N/A
SB 777-32-0058	SB 777-32-0058	001	01	B777 MLG Upper & Lower Hyd Support Repl	COMPLETED	2010.04.17	N/A

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-32-0069	SB 777-32-0069	000	00	LANDING GEAR - Main Landing Gear Extensi	COMPLETED	2008.01.17	N/A
SB 777-32-0069	SB 777-32-0069	001	00	LANDING GEAR - Main Landing Gear Extensi	COMPLETED	2008.01.17	N/A
SB 777-32-0070	SB 777-32-0070	000	01	L/H MLG DRAG&SIDE BRACE LOCK SPRING REPL	OPEN		N/A
SB 777-32-0070	SB 777-32-0070	001	01	R/H MLG DRAG&SIDE BRACE LOCK SPRING REPL	OPEN		N/A
SB 777-32-0079	SB 777-32-0079	000	00	B777 LMG Uplock Spring Replacement	OPEN		N/A
SB 777-32-0079	SB 777-32-0079	001	00	B777 RMG Uplock Spring Replacement	OPEN		N/A
SB 777-32-0084	SB 777-32-0084	000	00	L/H Extension and Retraction - Main Land	COMPLETED	2010.11.14	N/A
SB 777-32-0084	SB 777-32-0084	001	00	R/H Extension and Retraction - Main Land	COMPLETED	2010.11.14	N/A
SB 777-32-0086	SB 777-32-0086	000	02	B777 LMLG J501/503 Junction Box Bracket	OPEN		N/A
SB 777-32-0086	SB 777-32-0086	001	02	B777 RMLG J501/503 Junction Box Bracket	OPEN		N/A
SB 777-32-0088	SB 777-32-0088	000	01	Extension and Retraction - L/H Main Land	OPEN		N/A

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-32-0088	SB 777-32-0088	001	01	Extension and Retraction - R/H Main Land	OPEN		N/A
SB 777-32-0092	SB 777-32-0092	000	00	777 MLG Hydralic Fuse Assy Insp & Repl	OPEN		N/A
SB 777-32-0092	SB 777-32-0092	001	00	777 MLG Hydralic Fuse Assy Insp & Repl	OPEN		N/A
SB 777-32A0082	SB 777-32A0082	000	00	LANDING GEAR - L/H Main Landing Gear - S	COMPLETED(Initial Lub`)	2011.01.24	FAA 2012-19-10_000_00
SB 777-32A0082	SB 777-32A0082	001	00	LANDING GEAR - R/H Main Landing Gear - S	COMPLETED(Initial Lub`)	2011.01.24	FAA 2012-19-10_001_00
SB 777-32A0082	SB 777-32A0082	002	00	LANDING GEAR - R/H Main Landing Gear - S	COMPLETED(Inspection)	2012.12.05	e Incorporated by L/G Overhau
SB 777-32A0082	SB 777-32A0082	003	00	LANDING GEAR - L/H Main Landing Gear - S	COMPLETED(Inspection)	2012.12.05	e Incorporated by L/G Overhau
SB 777-33-0038	SB 777-33-0038	001	00	Passenger Compartment Illumination - Moo	COMPLETED	2012.09.24	N/A
SB 777-34A0191	SB 777-34A0191	000	00	NAVIGATION - Radio Altimeter System - Ra	OPEN		N/A
SB 777-34A0192	SB 777-34A0192	000	00	Radio Altimeter System - Low Range Radio	OPEN		N/A
SB 777-35-0019 R0	SB 777-35-0019	000	01	Oxygen Mask Flow Indicator Replacement	COMPLETED	2010.04.12	N/A

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-35-0020	SB 777-35-0020	000	01	B777 Oxygen Box Cable Routing Insp & Cor	COMPLETED	2010.04.15	N/A
SB 777-35A0023	SB 777-35A0023	000	00	Crew and Passenger Gaseous Oxygen System	COMPLETED	2010.01.13	N/A
SB 777-35A0027	SB 777-35A0027	000	01	OXYGEN - Crew Oxygen System - Low Pressu	COMPLETED	2012.12.06	N/A
SB 777-36-0016	SB 777-36-0016	001	00	PNEUMATIC - AIR SUPPLY AND CABIN PRESSUR	COMPLETED	2007.08.31	N/A
SB 777-36-0024 R0	SB 777-36-0024	000	00	PNEUMATIC -HPFAC, Bleed Air Tube, Change	COMPLETED	2007.09.30	N/A
SB 777-36-0024 R0	SB 777-36-0024	001	00	PNEUMATIC -HPFAC, Bleed Air Tube, Change	COMPLETED	2007.09.30	N/A
SB 777-36-0025	SB 777-36-0025	000	00	PNEUMATIC - Distribution - Air Supply an	COMPLETED	2007.05.17	N/A
SB 777-36-0027	SB 777-36-0027	000	00	PNEUMATIC - Engine Air Supply,Intermedia	COMPLETED	2007.09.16	N/A
SB 777-36-0027	SB 777-36-0027	001	00	PNEUMATIC - Engine Air Supply,Intermedia	COMPLETED	2007.09.16	N/A
SB 777-36-0028	SB 777-36-0028	000	00	B777 Engine Air Supply FAMV Change (RH)	COMPLETED	2010.04.20	N/A
SB 777-36-0028	SB 777-36-0028	001	00	B777 Engine Air Supply FAMV Change (LH)	COMPLETED	2010.10.18	N/A

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-36-0030	SB 777-36-0030	000	00	PNEUMATIC - Distribution - Air Supply an	COMPLETED	2011.05.20	N/A
SB 777-36-0031	SB 777-36-0031	000	01	PNEUMATIC - Distribution - Auxiliary Pow	COMPLETED	2012.12.08	N/A
SB 777-36A0026	SB 777-36A0026	000	01	AIR SUPPLY AND CABIN PRESSURE CONTROLLER	COMPLETED	2007.03.16	N/A
SB 777-36A0026	SB 777-36A0026	001	01	AIR SUPPLY AND CABIN PRESSURE CONTROLLER	COMPLETED	2007.05.17	N/A
SB 777-38-0032	SB 777-38-0032	000	00	Potable Water Press', Component Change	COMPLETED	2010.04.19	N/A
SB 777-38-0035	SB 777-38-0035	000	01	WATER/WASTE - Gray Water Drain System -	COMPLETED	2010.04.15	N/A
SB 777-38-0036	SB 777-38-0036	000	00	FWD Drain - Heated Tube Assembly Repl'	COMPLETED	2010.04.15	N/A
SB 777-49-0017	SB 777-49-0017	000	00	Auxiliary Power Unit (APU) - Power Cable	OPEN		N/A
SB 777-52-0039	SB 777-52-0039	000	00	DOORS - Equipment Compartment Exterior D	COMPLETED	2011.05.17	N/A
SB 777-52-0053	SB 777-52-0053	000	00	DOORS - Large Lower Lobe Cargo Door - Ro	OPEN		N/A
SB 777-52A0038	SB 777-52A0038	000	00	B777 Cargo Door LPF Latch Pin Bolt Insp	COMPLETED	2009.07.15	N/A

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-52A0038	SB 777-52A0038	000	01	LPF Latch Pin Retention Bolt Insp & Repl	COMPLETED	2010.10.10	N/A
SB 777-53-0050	SB 777-53-0050	000	00	B777 Underwing Accs' Door Rigging&Latch	COMPLETED	2010.04.12	N/A
SB 777-53-0058	SB 777-53-0058	000	01	Aerodynamic Fairing - Wing-to-Body Fairi	COMPLETED	2011.05.19	N/A
SB 777-53A0054	SB 777-53A0054	000	01	B777 Scribe Line Inspection (Part 6)	OPEN(Initial Insp`)		To Be Issued
SB 777-53A0054	SB 777-53A0054	001	01	B777 Scribe Line Inspection (Part 5)	OPEN(Initial Insp`)		To Be Issued
SB 777-53A0054	SB 777-53A0054	002	01	B777 Scribe Line Inspection (Part 4)	OPEN(Initial Insp`)		To Be Issued
SB 777-53A0054	SB 777-53A0054	003	01	B777 Scribe Line Inspection (Part 3)	OPEN(Initial Insp`)		To Be Issued
SB 777-53A0054	SB 777-53A0054	004	01	B777 Scribe Line Inspection (Part 1)	OPEN(Initial Insp`)		To Be Issued
SB 777-54-0023	SB 777-54-0023	000	00	ACELLES / PYLONS - Strut Drains and Seal	COMPLETED	2007.09.29	N/A
SB 777-54-0023	SB 777-54-0023	001	00	ACELLES / PYLONS - Strut Drains and Seal	COMPLETED	2007.09.29	N/A
SB 777-54-0025	SB 777-54-0025	000	00	AFT FARING LWR SPAR STIFFENER & CHORD CH	COMPLETED	2010.04.16	N/A

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-54-0026	SB 777-54-0026	000	01	L/H NACELLES/PYLONS - Aft Fairing Heat S	OPEN		N/A
SB 777-54-0026	SB 777-54-0026	001	01	R/H NACELLES/PYLONS - Aft Fairing Heat S	OPEN		N/A
SB 777-54-0027	SB 777-54-0027	000	00	B777 LH Eng FWD Strut Drain Line Insp	COMPLETED(Initial Insp`)	2012.12.04	7A-54-55-01-720-801
SB 777-54-0027	SB 777-54-0027	001	00	B777 RH Eng FWD Strut Drain Line Insp	COMPLETED(Initial Insp`)	2012.12.04	7A-54-55-01-720-801
SB 777-54-0028	SB 777-54-0028	000	00	L/H NACELLES / PYLONS - Nacelle Strut -	COMPLETED(Initial Insp`)	2013.03.13	FAA 2013-11-14_000_00
SB 777-54-0028	SB 777-54-0028	001	00	R/H NACELLES / PYLONS - Nacelle Strut -	COMPLETED(Initial Insp`)	2013.03.13	FAA 2013-11-14_001_00
SB 777-54A0024	SB 777-54A0024	000	00	No.1 Eng Strut Disconnect Box - Replace	COMPLETED(Terminating)	2010.04.18	N/A
SB 777-54A0024	SB 777-54A0024	001	00	No.2 Eng Strut Disconnect Box - Replace	COMPLETED(Terminating)	2010.04.19	N/A
SB 777-55A0015	SB 777-55A0015	000	02	B777 LH Elevator Acutator Fitting Replac	OPEN(Initial Insp`)		To Be Issued
SB 777-55A0015	SB 777-55A0015	001	02	B777 RH Elevator Acutator Fitting Insp	OPEN(Initial Insp`)		To Be Issued
SB 777-55A0016	SB 777-55A0016	000	01	B777 LH Elevator Actuator Fitting Replac	OPEN(Terminating Action, If Abnoraml Condition Found During Repeat Insp` as per SB 777-55A0015))		N/A

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-55A0016	SB 777-55A0016	001	01	B777 RH Elevator Actuator Fitting Replac	OPEN(Terminating Action, If Abnoraml Condition Found During Repeat Insp` as per SB 777-55A0015))		N/A
SB 777-55A0017	SB 777-55A0017	000	00	B777 Karon Lined Bushing Inspection	OPEN(Initial Insp`)		To Be Issued
SB 777-55A0018	SB 777-55A0018	000	02	B777 Horizontal Pivot Pin - Replacement	OPEN(Initial Insp`)		To Be Issued
SB 777-56-0002	SB 777-56-0002	000	00	WINDOWS - Flight Compartment Windows	COMPLETED	2008.11.18	N/A
SB 777-56-0004	SB 777-56-0004	000	00	B777 Window Middle Pane Crack Insp'	COMPLETED	2010.04.07	N/A
SB 777-56-0005	SB 777-56-0005	000	01	B777 Lh&RH NO.2 WINDOW MODIFICATION	COMPLETED	2010.04.14	N/A
SB 777-56-0005	SB 777-56-0005	000	02	B777 Lh&RH NO.2 WINDOW MODIFICATION	COMPLETED	2012.12.09	N/A
SB 777-57-0058	SB 777-57-0058	000	00	MLG Hanger Link Pin Replacement	COMPLETED	2010.04.16	N/A
SB 777-57-0063	SB 777-57-0063	000	00	Wing Center Section Sealant Repair	COMPLETED	2010.04.17	N/A
SB 777-57-0083	SB 777-57-0083	000	01	B777 PIP_Vortex Generator Replacement	COMPLETED	2012.12.10	N/A
SB 777-57-0089	SB 777-57-0089	000	01	WINGS - Leading Edge and Leading Edge De	OPEN		N/A

SB No.	MEO No.	MEO Part	MEO Rev.	Description	Status	Accomplished Date	Repeat Insp` No
SB 777-57-0089	SB 777-57-0089	001	01	WINGS - Leading Edge and Leading Edge De	OPEN		N/A
SB 777-57A0059	SB 777-57A0059	000	00	WINGS - Spars - Fuel Tank Fasteners Cap	COMPLETED	2010.04.12	N/A
SB 777-57A0059	SB 777-57A0059	001	00	WINGS - Spars - Fuel Tank Fasteners Cap	COMPLETED	2010.04.12	N/A
SB 777-57A0064	SB 777-57A0064	000	00	LH WING L/E SLAT DOWNSTOP AND CAN INSP'	COMPLETED	2009.06.17	N/A
SB 777-57A0064	SB 777-57A0064	001	00	RH WING L/E SLAT DOWNSTOP AND CAN INSP'	COMPLETED	2009.06.17	N/A
SB 777-57A0064	SB 777-57A0064	000	01	LH WING L/E SLAT DOWNSTOP AND CAN INSP'	OPEN		N/A
SB 777-57A0064	SB 777-57A0064	001	01	RH WING L/E SLAT DOWNSTOP AND CAN INSP'	OPEN		N/A
SB 777-57A0087	SB 777-57A0087	000	01	WINGS - Center Wing - Beams - Spanwise B	OPEN(Initial Insp`)		FAA 2012-08-09_000_00
SB 777-57A0090	SB 777-57A0090	000	01	WINGS - Landing Gear Attach Fittings and	OPEN		N/A
SB 777-57A0090	SB 777-57A0090	0001	01	WINGS - Landing Gear Attach Fittings and	OPEN		N/A

8-2-1.Engine (P222240) SB Status

SB No.	REV	Description	P/N	S/N	Notification	AAR Status	Accomplished Date	Remark
SB 120EGS11I-24-6	00	SB 120EGS11I-24-6 :IDG-REPLACE INPUT S	767146A	1447	144700241222	COMPLETED	2004-11-24	Refer to the 120EGS11I-24-6.
SB 015864-79-3	00	Replace Lube Spur Matched Gearset	51R159	CNMUAJ0221	3100204767	OPEN		Supersedes SB PW4G-112-79-32
SB 07P95-75-1075	02	Rework Pneumatic Relay Valve (PRV), PN 0	52U880	CSGCB19133	3100308401	OPEN		Supersedes PW4G-112-75-50
SB 1110V0700-75-1101	00	TVBC Valve - New Indicator Assembly	55H418	CSGCBJ6262	3100238329	OPEN		
SB 1110V0700-75-1101	00	TVBC Valve - New Indicator Assembly	55H418	CSGCBR5899	3100238335	OPEN		
SB 1110V0700-75-1101	00	TVBC Valve - New Indicator Assembly	55H418	CSGCB20600	3100238368	OPEN		
SB 120EGS11I-24-11	01	INTEGRATED DRIVE GENERATOR - REPLACE BEA	767146A	1447	3100344116	OPEN		
SB 120EGS11I-24-7	01	Electrical Power - Integrated Drive Gene	767146A	1447	3100229688	OPEN		
SB 120EGS11I-24-8	01	ELECTRICAL POWER - INTEGRATED DRIVE GENE	767146A	1447		OPEN		
SB 20VSCFGEN01-24-12	01	SB 20VSCFGEN01-24-12 :VSCF GENERATOR-R	1701768	0980	98000241198	COMPLETED	2002-09-16	
SB 20VSCFGEN01-24-15	01	SB 20VSCFGEN01-24-15 :VSCF GENERATOR-R	1701768	0980	98000241206	COMPLETED	2002-09-16	
SB 20VSCFGEN01-24-18	00	Electrical Power - VSCF Generator - Repl	1701768	0980	3100110548	COMPLETED	2012-06-15	
SB 20VSCFGEN01-24-20	00	ELECTRICAL POWER - VSCF GENERATOR - REPL	1701768	0980	3100234495	COMPLETED	2012-06-15	
SB 3215302-36-1885	00	PNEUMATIC - HIGH PRESSURE SHUTOFF/PRESSU	3215302-5	2130	3100321494	COMPLETED	2013-02-23	
SB 45731-73-004	00	Introduction of Servo Fuel Heater	51R148	CSGCBY4630	3100211112	OPEN		
SB 5930638-75-234	00	LPT Case Cooling Valve - Modification	5930638-102	0824	3100380041	OPEN		
SB 5930639-75-232	00	HPT Case Cooling Valve - Modification	50U145	0758	3100240878	OPEN		
SB 972424-29-04	00	SB 972424-29-04 :HYD' PUMP ASSY - ACTU	972952	MX637000	3100007158	OPEN		
SB EEC170-1-73-28	00	ENGINE FUEL AND CONTROL - EEC170-1 ELECT	822830-15	4171-0472	3100319100	OPEN		
SB GTA41-1-75-14	00	AIR - GTA41-1 STATOR VANE ACTUATOR - HEL	56H267	F63477	3100265632	OPEN		
SB GTA41-1-75-15	00	AIR - GTA41-1 STATOR VANE ACTUATOR - INC	56H267	F63477	3100265806	OPEN		

SB No.	REV	Description	P/N	S/N	Notification	AAR Status	Accomplished Date	Remark
SB GTA42-2-75-11	00	AIR - GTA42-2 INTERCOMPRESSOR (2.5) BLEE	52G377	F63531	3100266115	OPEN		
SB GTA49-1-75-9	00	AIR - GTA49-1 TURBINE CASE COOLING VALVE	54H217	F63436	3100266156	OPEN		
SB JFC131-4-73-17	00	JFC131-4 FUEL METERING UNIT - INCORPORA	808800-1	CSGCB33939	3100362776	OPEN		
SB PW4G-112-72-119	02	ENGINE . AIRSEAL, DIFFUSER . DIAMETER ME	PW4090	P222240		OPEN		104347803 (Internal Document)
SB PW4G-112-72-254	04	BLADE, 7TH STAGE HPC - BORESCOPE INSPECT	PW4090	P222240		REPETITIVE		GTL 22791 / GTL 22792
SB PW4G-112-72-255	04	BLADE, 8TH STAGE HPC - BORESCOPE INSPECT	PW4090	P222240		REPETITIVE		GTL 22791/2 GTL 22792
SB PW4G-112-72-26	07	VANE, 2ND STAGE HIGH PRESSURE TURBINE (H	PW4090	P222240		REPETITIVE		GTL No. 22781, 22782
SB PW4G-112-72-260	00	SB 72-260 HPT S2 BLD - INSP FOR SCC UNDE	PW4090	P222240		REPETITIVE		
SB PW4G-112-72-264	02	SB72-264R1 HPC DRUM ROTOR-SHOP REPEAT FP	PW4090	P222240	3100011118	REPETITIVE		AD 2012-18-17
SB PW4G-112-72-283 R0	01	ENGINE . BOLT, BRACKET ASSEMBLY, EXHAUST	PW4090	P222240	3100027985	OPEN		
SB PW4G-112-72-285 R0	01	VANE ASSY - 3RD & 4TH STG LPT - MAGNETOS	PW4090	P222240	3100032485	OPEN		
SB PW4G-112-72-287	00	VANE ASSEMBLY, LOW PRESSURE TURBINE (LPT	PW4090	P222240	3100077111	OPEN		
SB PW4G-112-72-288	01	ENGINE . BLADE, 2ND STAGE HIGH PRESSURE	PW4090	P222240	3100082582	OPEN		
SB PW4G-112-72-289	01	ENGINE . BLADE, 1ST STAGE HIGH PRESSURE	PW4090	P222240	3100089273	COMPLETED	2009-02-09	
SB PW4G-112-72-291	00	ENGINE . BLADES, 5TH THRU 15TH STAGE HIG	PW4090	P222240	3100088544	OPEN		
SB PW4G-112-72-292	00	ENGINE . ROTOR AND STATOR ASSEMBLY, HIGH	PW4090	P222240	3100135086	COMPLETED	2009-02-09	
SB PW4G-112-72-292	01	ENGINE - ROTOR AND STATOR ASSEMBLY, HIGH	PW4090	P222240	3100300146	OPEN		
SB PW4G-112-72-293	00	ENGINE . SEAL, DIFFUSER AIR (INNER) . ED	PW4090	P222240	3100265230	COMPLETED	2009-02-09	
SB PW4G-112-72-294	00	ENGINE . BEARING, NO. 1 BALL . MODIFICAT	PW4090	P222240	3100136307	COMPLETED	2013-04-22	
SB PW4G-112-72-295	00	SEAL ASSEMBLIES, NO. 3 BEARING, FRONT AN	PW4090	P222240	3100113516	COMPLETED	2009-02-09	
SB PW4G-112-72-296	01	ENGINE .VANES, INLET THRU 14TH STAGE HIG	PW4090	P222240	3100114450	OPEN		
SB PW4G-112-72-297	02	BLADE ASSEMBLIES, STAGE 3, 4, 5, AND 6 L	PW4090	P222240	3100223672	COMPLETED	2013-04-22	
SB PW4G-112-72-298	01	ENGINE . CLEVIS, ROD END, 2.5 BLEED VALV	PW4090	P222240	3100137202	COMPLETED	2009-02-09	

SB No.	REV	Description	P/N	S/N	Notification	AAR Status	Accomplished Date	Remark
SB PW4G-112-72-300	00	ENGINE .VANE CLUSTER ASSEMBLY, 9TH STAGE	PW4090	P222240		COMPLETED	2009-02-09	Terminated by SB 72-301
SB PW4G-112-72-301	00	ENGINE .VANE CLUSTER ASSEMBLIES, 9TH STA	PW4090	P222240	3100152232	COMPLETED	2009-02-09	
SB PW4G-112-72-303	01	ENGINE . HIGH PRESSURE COMPRESSOR (HPC)	PW4090	P222240	3100187074	OPEN		
SB PW4G-112-72-306	01	ENGINE - SEAL ASSEMBLIES, GEARBOX - REPL	PW4090	P222240	3100236997	COMPLETED	2013-04-22	
SB PW4G-112-72-307	00	STRAINER ASSEMBLY, GEARBOX - REPLACEMEN	PW4090	P222240	3100239437	COMPLETED	2013-04-22	
SB PW4G-112-72-308	00	ENGINE - TIEROD - LOW PRESSURE COMPRESSO	PW4090	P222240	3100205244	COMPLETED	2013-04-22	
SB PW4G-112-72-310	01	ENGINE - FACE SEAL ASSEMBLY, NO. 1.5 BEA	PW4090	P222240	3100231990	COMPLETED	2013-04-22	
SB PW4G-112-72-311	00	BLADE, 7TH AND 8TH STAGE HIGH PRESSURE C	PW4090	P222240	3100265851	OPEN		
SB PW4G-112-72-313	00	ENGINE - HEAT SHIELDS, NO. 3 BEARING TUB	PW4090	P222240	3100297639	OPEN		
SB PW4G-112-72-315	01	ENGINE - SEAL, OUTER AIR (OAS) AND INSUL	PW4090	P222240	3100300120	COMPLETED	2013-04-22	
SB PW4G-112-72-316	00	BLADE, 1ST STAGE HIGH PRESSURE TURBINE (	PW4090	P222240	3100313182	COMPLETED	2013-04-22	
SB PW4G-112-72-317	01	BLADE, 7TH AND 8TH STAGE HIGH PRESSURE C	PW4090	P222240	3100314318	OPEN		
SB PW4G-112-72-318	00	SEAL HOUSING - NO. 4 BEARING, TURBINE EX	PW4090	P222240	3100313644	OPEN		
SB PW4G-112-72-319	00	BLADE ASSEMBLY, 2ND STAGE HIGH PRESSURE	PW4090	P222240	3100319164	OPEN		
SB PW4G-112-72-320	00	SHROUD SET, INLET THRU 7TH STAGE, HPC -	PW4090	P222240	3100320030	COMPLETED	2013-04-22	
SB PW4G-112-72-321	00	STATOR ASSEMBLY, 4TH STAGE LOW PRESSURE	PW4090	P222240	3100326005	OPEN		
SB PW4G-112-72-322	00	FACE SEAL ASSEMBLY, NO. 1.5 BEARING - MO	PW4090	P222240	3100328578	COMPLETED	2013-04-22	
SB PW4G-112-72-324	00	VANES, 5TH THRU 7TH STAGE HPC - GIVE A P	PW4090	P222240	3100350990	OPEN		
SB PW4G-112-72-325	00	FAN EXIT CASE ASSEMBLY - SUPPLY A FAN EX	PW4090	P222240	3100339261	COMPLETED	2013-04-22	
SB PW4G-112-72-326	00	BLADE - 1ST STAGE HPT - SUPPLY A BLADE T	PW4090	P222240	3100343782	COMPLETED	2013-04-22	
SB PW4G-112-72-328	00	LOCK AND KEY, 2ND STAGE HPT - REPLACEMEN	PW4090	P222240	3100371825	COMPLETED	2013-04-22	
SB PW4G-112-72-329	00	RING SEGMENT AND VANE CLUSTER ASSEMBLY,	PW4090	P222240	3100392102	OPEN		
SB PW4G-112-73-40	00	To provide a new electrical connector en	PW4090	P222240	3100091562	OPEN		

SB No.	REV	Description	P/N	S/N	Notification	AAR Status	Accomplished Date	Remark
SB PW4G-112-73-42	01	ENGINE FUEL AND CONTROL . EEC170-1 ELECT	822830-15	4171-0472	3100192808	OPEN		
SB PW4G-112-73-44	01	W0623 Modification To Prevent Chafing	PW4090	P222240	3000831103	COMPLETED	2010-11-17	
SB PW4G-112-74-6	00	IGNITION EXCITER, HEATSHIELDS, AND BRACK	PW4090	P222240	3100372753	OPEN		
SB PW4G-112-75-23	02	BRACKET ASSY - FPI FOR CRACKS		P222240	3100405271	REPETITIVE		
SB PW4G-112-75-46	00	AIR-BLEED AIR VALVE - MODIFY VALVE TO REP	58H107	CSGCB08155		N/A		
SB PW4G-112-75-46	00	AIR-BLEED AIR VALVE - MODIFY VALVE TO REP	58H107	CSGCB08157		N/A		
SB PW4G-112-75-51	02	BLEED VALVES, 2.9 START AND STABILITY .	58H107	CSGCB08155	3100190851	COMPLETED	2009-02-09	
SB PW4G-112-75-51	02	BLEED VALVES, 2.9 START AND STABILITY .	58H107	CSGCB08157	3100190852	COMPLETED	2009-02-09	
SB PW4G-112-75-52	00	AIR . SOLENOID, HPC SECONDARY FLOW CONTR	PW4090	P222240	3100052616	OPEN		
SB PW4G-112-77-11	00	ENGINE INDICATING . MANIFOLD AND TUBE AS	PW4090	P222240	3100265257	COMPLETED	2013-04-22	
SB PW4G-112-79-33	00	OIL . ELBOW, MAIN OIL FILTER TUBE (LP01)	PW4090	P222240	3100226887	COMPLETED	2013-04-22	
SB PW4G-112-A72-242	03	SB A72-242 15TH STG HPC DISK-BSI FOR CRA	PW4090	P222240		REPETITIVE		AD Card FAA 2002-09-01
SB PW4G-112-A72-268	04	ENGINE . BLADE ASSEMBLY, 1ST STAGE, LOW	PW4090	P222240	3100236825	REPETITIVE		Third party full accomp.
SB PW4G-112-A72-280	02	ENGINE . SEAL ASSEMBLY, BRUSH, HIGH PRES	PW4090	P222240		COMPLETED	2006-09-29	Not applicable - FMP prohibits used New Brush Seal
SB PW4G-112-A72-330	00	PW4090 T2 Air-seal Inspection (ECI)	PW4090	P222240	3001279308	OPEN		
SB PW4G-112-A79-11	02	OIL - PERIODIC INSP #3,4 BEARING TUBES	PW4090	P222240		REPETITIVE		GTL 22823 / 22824

8-2-2. Engine (P222186) SB Status

SB No.	REV	Description	P/N	S/N	AAR Status	Accomplished Date	Remark
SB 3399100-36-1828	00	MEP/SB 3399100-36-1828/00/000, MEP/SB 3399100-36-1828/00/000, MEP/SB 3399100-36-1828/00/000, MEP/SB 3399100-36-1828/00/000	3399100-30	1056	OPEN		
SB 015864-79-3	00	Replace Lube Spur Matched Gearset	51R159	H0983	OPEN		Supersedes PW4G-112-79-32
SB 07P95-75-1075	02	Rework Pneumatic Relay Valve (PRV), PN 0	52G435	CSGCBN5387	OPEN		Supersedes PW4G-112-75-50
SB 1110V0700-75-1101	00	TVBC Valve - New Indicator Assembly	55H418	CSGCBR5904	OPEN		
SB 1110V0700-75-1101	00	TVBC Valve - New Indicator Assembly	55H418	CSCBR5907	OPEN		
SB 1110V0700-75-1101	00	TVBC Valve - New Indicator Assembly	1110V0700-5	3003	COMPLETED	2012-07-20	
SB 1110V0700-75-1101	00	TVBC Valve - New Indicator Assembly	1110V0700-5	CSGCB20601	COMPLETED	2012-04-28	
SB 120EGS11I-24-11	01	INTEGRATED DRIVE GENERATOR - REPLACE BEA	767146A	AAAG002043	OPEN		
SB 120EGS11I-24-8	01	ELECTRICAL POWER - INTEGRATED DRIVE GENE	767146A	AAAG002043	OPEN		
SB 12698 24-00	00	INTEGRATED DRIVE GENERATOR (IDG) AIR OIL	50G479	HL7742_1646	OPEN		
SB 15622 79-00	00	OIL - ENGINE AIR OIL COOLER - REPLACEMEN	51G859	CSGCBR3268	OPEN		
SB 20VSCFGEN01-24-19	00	Electrical Power - VSCF Generator - Repl	1701768	20794	OPEN		
SB 20VSCFGEN01-24-20	00	ELECTRICAL POWER - VSCF GENERATOR - REPL	1701768	20794	OPEN		
SB 3215302-36-1885	00	PNEUMATIC - HIGH PRESSURE SHUTOFF/PRESSU	3215302-4	3281	OPEN		
SB 45731-73-004	00	Introduction of Servo Fuel Heater	51R148	CSGCBR6250	OPEN		
SB 540-0001-75-1066	00	SB 540-0001-75-1066 :INSPECTION BLEED	58H107	CSGCBR4864C	REPETITIVE		
SB 540-0001-75-1066	00	SB 540-0001-75-1066 :INSPECTION BLEED	58H107	CSGCBR4866C	REPETITIVE		
SB 5930638-75-234	00	LPT Case Cooling Valve - Modification	57T496	0614	OPEN		
SB 5930639-75-232	00	HPT Case Cooling Valve - Modification	50U145	0618	OPEN		
SB 777-78-0052	02	EXHAUST (PW40011 EA0 ENGINES) - Exhaust	314W3520-35	W462282	OPEN		
SB 972424-29-04	00	SB 972424-29-04 :HYD' PUMP ASSY - ACTU	972952	MX636993	COMPLETED	2009-04-15	
SB EEC170-1-73-28	00	ENGINE FUEL AND CONTROL - EEC170-1 ELECT	822830-15	HL7742_1644	OPEN		

SB No.	REV	Description	P/N	S/N	AAR Status	Accomplished Date	Remark
SB GTA41-1-75-14	00	AIR - GTA41-1 STATOR VANE ACTUATOR - HEL	819810-1	F64022	OPEN		
SB GTA41-1-75-15	00	AIR - GTA41-1 STATOR VANE ACTUATOR - INC	819810-1	F64022	OPEN		
SB GTA42-2-75-11	00	AIR - GTA42-2 INTERCOMPRESSOR (2.5) BLEE	814890-2	F9108	OPEN		
SB GTA49-1-75-9	00	AIR - GTA49-1 TURBINE CASE COOLING VALVE	54H217	F54895	OPEN		
SB JFC131-4-73-17	00	JFC131-4 FUEL METERING UNIT - INCORPORA	808800-1	F54262	OPEN		
SB PW4G-100-75-47	00	AIR - VALVE, PNEUMATIC RELAY . REPLACEME	52G435	CSGCBN5387	COMPLETED	2006-08-31	
SB PW4G-112-72-119	02	ENGINE . AIRSEAL, DIFFUSER . DIAMETER ME	PW4090	P222186	OPEN		104347803 (Internal Document)
SB PW4G-112-72-189	02	HUB ASS'Y-LPC-INSP'& IDENTIFICATION OF N	PW4090	P222186	COMPLETED	2002-08-16	
SB PW4G-112-72-254	04	BLADE, 7TH STAGE HPC - BORESCOPE INSPECT	PW4090	P222186	REPETITIVE		GTL 22791 / GTL 22792
SB PW4G-112-72-255	04	BLADE, 8TH STAGE HPC - BORESCOPE INSPECT	PW4090	P222186	REPETITIVE		GTL 22791 / GTL 22792
SB PW4G-112-72-256	00	INSPECT FOR BOLT HOLE CRACKS ON FLANGE J	PW4090	P222186	REPETITIVE		
SB PW4G-112-72-259	01	PW4G-112-72-259 THRUST BEARING WASHER RE	PW4090	P222186	COMPLETED	2009-05-04	
SB PW4G-112-72-26	07	VANE, 2ND STAGE HIGH PRESSURE TURBINE (H	PW4090	P222186	REPETITIVE		(GTL No. 22781, 22782)
SB PW4G-112-72-260	00	INSP' STRESS CORROSION CRACKING UNDER RO	PW4090	P222186	REPETITIVE		
SB PW4G-112-72-260	00	SB 72-260 HPT S2 BLD - INSP FOR SCC UNDE	PW4090	P222186	REPETITIVE		
SB PW4G-112-72-264	02	DISK, ASSY OF, HPC (HIGH PRESSURE COMPRE	PW4090	P222186	REPETITIVE		AD 2012-18-17
SB PW4G-112-72-269	00	SB 72-269 BOLT, E FLANGE - REPLACEMENT O	PW4090	P222186	COMPLETED	2006-08-31	
SB PW4G-112-72-271	01	SB72-271(LLP ATTRITION)HPT S2 AIR SEAL A	PW4090	P222186	COMPLETED	2006-08-31	
SB PW4G-112-72-273	00	SB 72-273, NEW COMP STATOR ASSY, 4TH STG	PW4090	P222186	COMPLETED	2006-08-31	
SB PW4G-112-72-274	00	SB 72-274 CASE, REAR HPC - MODIFICATION	PW4090	P222186	OPEN		
SB PW4G-112-72-275	00	SB72-275(ATTRITION-LLP)HPT S1 ROTOR ASSY	PW4090	P222186	OPEN		
SB PW4G-112-72-276	01	ENGINE . ROTOR ASSEMBLY, 2ND STAGE TURBI	PW4090	P222186	COMPLETED	2006.08.31	
SB PW4G-112-72-277	00	BEARING, SLEEVE, HIGH PRESSURE COMPRESSO	PW4090	P222186	COMPLETED	2009-05-04	
SB PW4G-112-72-282	01	ENGINE - HUB ASSEMBLY, FRONT TURBINE INS	PW4090	P222186	COMPLETED	2006.08.31	
SB PW4G-112-72-283 R0	00	ENGINE . BOLT, BRACKET ASSEMBLY, EXHAUST	PW4090	P222186	OPEN	2009-07-17	

SB No.	REV	Description	P/N	S/N	AAR Status	Accomplished Date	Remark
SB PW4G-112-72-285 R0	01	VANE ASSY - 3RD & 4TH STG LPT - MAGNETOS	PW4090	P222186	OPEN		
SB PW4G-112-72-285 R0	00	VANE ASSY - 3RD & 4TH STG LPT - MAGNETOS	PW4090	P222186	OPEN		
SB PW4G-112-72-287	01	VANE ASSEMBLY, LOW PRESSURE TURBINE (LPT	PW4090	P222186	OPEN		
SB PW4G-112-72-288	00	ENGINE . BLADE, 2ND STAGE HIGH PRESSURE	PW4090	P222186	OPEN		
SB PW4G-112-72-289	01	ENGINE . BLADE, 1ST STAGE HIGH PRESSURE	PW4090	P222186	OPEN		
SB PW4G-112-72-290	00	ENGINE . HUB ASSEMBLY, FRONT TURBINE . M	PW4090	P222186	COMPLETED	2009-05-04	
SB PW4G-112-72-291	00	ENGINE . BLADES, 5TH THRU 15TH STAGE HIG	PW4090	P222186	OPEN		
SB PW4G-112-72-292	00	ENGINE . ROTOR AND STATOR ASSEMBLY, HIGH	PW4090	P222186	COMPLETED	2009-05-04	
SB PW4G-112-72-292	01	ENGINE - ROTOR AND STATOR ASSEMBLY, HIGH	PW4090	P222186	COMPLETED	2009-05-04	
SB PW4G-112-72-293	00	ENGINE . SEAL, DIFFUSER AIR (INNER) . ED	PW4090	P222186	OPEN		
SB PW4G-112-72-294	00	ENGINE . BEARING, NO. 1 BALL . MODIFICAT	PW4090	P222186	OPEN		
SB PW4G-112-72-295	00	SEAL ASSEMBLIES, NO. 3 BEARING, FRONT AN	PW4090	P222186	COMPLETED	2009-05-04	
SB PW4G-112-72-296	01	ENGINE .VANES, INLET THRU 14TH STAGE HIG	PW4090	P222186	OPEN		
SB PW4G-112-72-297	02	BLADE ASSEMBLIES, STAGE 3, 4, 5, AND 6 L	PW4090	P222186	OPEN		
SB PW4G-112-72-298	00	ENGINE . CLEVIS, ROD END, 2.5 BLEED VALV	PW4090	P222186	COMPLETED	2009-05-04	
SB PW4G-112-72-301	00	ENGINE .VANE CLUSTER ASSEMBLIES, 9TH STA	PW4090	P222186	OPEN		
SB PW4G-112-72-303	01	ENGINE . HIGH PRESSURE COMPRESSOR (HPC)	PW4090	P222186	OPEN		
SB PW4G-112-72-306	01	ENGINE - SEAL ASSEMBLIES, GEARBOX - REPL	PW4090	P222186	OPEN		
SB PW4G-112-72-307	00	STRAINER ASSEMBLY, GEARBOX - REPLACEMEN	PW4090	P222186	OPEN		
SB PW4G-112-72-308	00	ENGINE - TIEROD - LOW PRESSURE COMPRESSO	PW4090	P222186	OPEN		
SB PW4G-112-72-310	01	ENGINE - FACE SEAL ASSEMBLY, NO. 1.5 BEA	PW4090	P222186	OPEN		
SB PW4G-112-72-310	00	ENGINE - FACE SEAL ASSEMBLY, NO. 1.5 BEA	PW4090	P222186	OPEN		
SB PW4G-112-72-311	00	BLADE, 7TH AND 8TH STAGE HIGH PRESSURE C	PW4090	P222186	OPEN		
SB PW4G-112-72-313	00	ENGINE - HEAT SHIELDS, NO. 3 BEARING TUB	PW4090	P222186	OPEN		
SB PW4G-112-72-315	00	ENGINE - SEAL, OUTER AIR (OAS) AND INSUL	PW4090	P222186	OPEN		

SB No.	REV	Description	P/N	S/N	AAR Status	Accomplished Date	Remark
SB PW4G-112-72-316	00	BLADE, 1ST STAGE HIGH PRESSURE TURBINE (	PW4090	P222186	OPEN		
SB PW4G-112-72-317	00	BLADE, 7TH AND 8TH STAGE HIGH PRESSURE C	PW4090	P222186	OPEN		
SB PW4G-112-72-318	00	SEAL HOUSING - NO. 4 BEARING, TURBINE EX	PW4090	P222186	OPEN		
SB PW4G-112-72-319	00	BLADE ASSEMBLY, 2ND STAGE HIGH PRESSURE	PW4090	P222186	OPEN		
SB PW4G-112-72-320	00	SHROUD SET, INLET THRU 7TH STAGE, HPC -	PW4090	P222186	OPEN		
SB PW4G-112-72-321	00	STATOR ASSEMBLY, 4TH STAGE LOW PRESSURE	PW4090	P222186	OPEN		
SB PW4G-112-72-322	00	FACE SEAL ASSEMBLY, NO. 1.5 BEARING - MO	PW4090	P222186	OPEN		
SB PW4G-112-72-324	00	VANES, 5TH THRU 7TH STAGE HPC - GIVE A P	PW4090	P222186	OPEN		
SB PW4G-112-72-325	00	FAN EXIT CASE ASSEMBLY - SUPPLY A FAN EX	PW4090	P222186	OPEN		
SB PW4G-112-72-326	00	BLADE - 1ST STAGE HPT - SUPPLY A BLADE T	PW4090	P222186	OPEN		
SB PW4G-112-72-328	00	LOCK AND KEY, 2ND STAGE HPT - REPLACEMEN	PW4090	P222186	OPEN		
SB PW4G-112-72-329	00	RING SEGMENT AND VANE CLUSTER ASSEMBLY,	PW4090	P222186	OPEN		
SB PW4G-112-73-16	02	FM02,FM08 FUEL SUPPLY TUBE-INSP'OF CLAMP	PW4090	P222186	COMPLETED	2002-08-16	
SB PW4G-112-73-36	0	WIRING HARNESS, W0626 - MODIFICATION OF	PW4090	P222186	COMPLETED	2006.08.31	
SB PW4G-112-73-38	00	SB 73-38 EEC -INCORPORATION OF SCN11 SOF	PW4090	P222186	COMPLETED	2005-11-16	
SB PW4G-112-73-40	00	To provide a new electrical connector en	PW4090	P222186	OPEN		
SB PW4G-112-73-42	01	ENGINE FUEL AND CONTROL . EEC170-1 ELECT	822830-15	HL7742_1644	OPEN		
SB PW4G-112-73-44	00	W0623 Modification To Prevent Chafing	PW4090	P222186	COMPLETED	2010-08-15	
SB PW4G-112-74-6	00	IGNITION EXCITER, HEATSHIELDS, AND BRACK	PW4090	P222186	OPEN		
SB PW4G-112-75-23	02	INSTR' TO DO A FLUORESCENT PENETRANT INS	PW4090	P222186	REPETITIVE		
SB PW4G-112-75-46	00	SB 75-46 MODIFY 2.9 BLEED VALVE TO REPLA	58H107	CSGCBR4864C	COMPLETED	2006.08.31	
SB PW4G-112-75-46	00	SB 75-46 MODIFY 2.9 BLEED VALVE TO REPLA	58H107	CSGCBR4866C	COMPLETED	2006.08.31	
SB PW4G-112-75-47	00	SB 75-47 TCC ACT' W/ IMPROVED PISTON SUR	PW4090	P222186	COMPLETED	2006.08.31	
SB PW4G-112-75-50	00	VALVE, PNEU' RELAY - REPLACE OR MODIFY	PW4090	P222186	OPEN		
SB PW4G-112-75-51	02	BLEED VALVES, 2.9 START AND STABILITY .	58H107	CSGCBR4864C	COMPLETED	2006.08.31	

SB No.	REV	Description	P/N	S/N	AAR Status	Accomplished Date	Remark
SB PW4G-112-75-51	02	BLEED VALVES, 2.9 START AND STABILITY .	58H107	CSGCBR4866C	COMPLETED	2006.08.31	
SB PW4G-112-75-52	00	BLEED VALVES, 2.9 START AND STABILITY .	PW4090	P222186	OPEN		
SB PW4G-112-77-11	00	ENGINE INDICATING . MANIFOLD AND TUBE AS	PW4090	P222186	OPEN		
SB PW4G-112-77-7	00	TRAP CONDENSATION-INSP'& CLEANING FOR BL	PW4090	P222186	COMPLETED	2002-08-16	
SB PW4G-112-77-9	00	SB 77-9 E-FLANGE ACCELEROMETER - REPLACE	82U864	CSGCB05090	COMPLETED	2006.08.31	
SB PW4G-112-79-27	01	SB 79-27 NO.3 BRG BREATHER TUBE-INSTALL	PW4090	P222186	OPEN	2010-06-08	
SB PW4G-112-79-28	00	SB PW4G-112-79-28 NEW PACKING FOR OIL TA	PW4090	P222186	OPEN		
SB PW4G-112-79-33	00	OIL . ELBOW, MAIN OIL FILTER TUBE (LP01)	PW4090	P222186	OPEN		
SB PW4G-112-80-4	00	SB 80-4 REPLACEMENT OF WITH A MORE DURAB	PW4090	P222186	COMPLETED	2006.08.31	
SB PW4G-112-80-6 (PS700-80-6507)	02	STARTING .VALVE, STARTER AIR . MODIFICAT	1001100-3	200109217	COMPLETED	2010.02.28	
SB PW4G-112-A72-233	02	INSP' FOR CRACKS & CYCLE LIMIT REDUCTION	PW4090	P222186	COMPLETED	2006.08.31	Terminated by SB 72-271
SB PW4G-112-A72-242	03	BSI FOR CRACKS IN FRONT RAIL OF BLADE LO	PW4090	P222186	REPETITIVE		AD Card FAA 2002-09-01
SB PW4G-112-A72-257	02	SB A72-257 INSPECTION OF NO. 3 BRG OIL V	PW4090	P222186	COMPLETED	2003-08-23	Terminated by SB A72-270
SB PW4G-112-A72-268	04	ENGINE . BLADE ASSEMBLY, 1ST STAGE, LOW	PW4090	P222186	REPETITIVE		
SB PW4G-112-A72-270	00	REPL' OR MOD' OF THE FRONT & REAR FACE S	PW4090	P222186	COMPLETED	2006.08.31	
SB PW4G-112-A72-270	00	SB A72-270 SEAL ASSY, No.3 BRG-REPLACE O	PW4090	P222186	COMPLETED	2006.08.31	
SB PW4G-112-A72-280	02	ENGINE . SEAL ASSEMBLY, BRUSH, HIGH PRES	PW4090	P222186	COMPLETED	2006-09-29	Not applicable - FMP prohibits used New Brush Seal
SB PW4G-112-A72-330	00	PW4090 T2 Air-seal Inspection (ECI)	PW4090	P222186	OPEN		
SB PW4G-112-A79-11	02	OIL - PERIODIC INSP #3,4 BEARING TUBES	PW4090	P222186	REPETITIVE		GTL 22823 / 22824